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Systems of Basis Functions for Calculation of Three-Dimensional Fluid Flows in Cylindrical Containers with the Galerkin Spectral Method ガラーキン法を用いた円筒容器内 3 次元流れの計算のための試行関数系について

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1. INTRODUCTION

In several experimental and numerical investigations of fluid flows in circular cylindrical domains a threshold from axisymmetric to asymmetric motion was reported (see for example, ref. [1-6]). Numerical investigation of such thresholds is usually carried out with direct straightforward solution of 3D equations with use of 2D flow as initial state. On the other hand the use of stability theory may provide more precise values of critical parameters as well as valid values of unstable azimuthal modes, but the instability analysis is carried out, as a rule, for problems with analytically known initial state. One of the simplest examples is Rayleigh-Benard convective instability in vertical cylinders heated from below, for which several experimental and theoretical investigations reported a theshold from quiescent state to 3D convective flow if the ratio height/radius of the cylinder is larger than 1 (see ref. [1, 2, 5]).

Application of linear stability analysis to investigation of instability of fluid flows leads to an eigenvalue problem of very high order. As it was shown in [7] for convective problems in rectangular regions the order of the eigenvalue problem (equal to the number of scalar modes used by a numerical method) may be reduced if the Galerkin method with divergent-free basis satisfying all the boundary conditions is applied. The present paper deals with a possibility to construct such a bases for numerical solution of hydrodynamical problems in confined circular cylinders. Problem of Rayleigh-Benard instability in a cylinder heated from below is used to illustrate the applicability of the constructed bases.

2. SYSTEM OF THE BASIS FUNCTIONS

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The main difficulty in formulation of the Galerkin spectral method for calculation of 3D flows in cylindrical as well as in spherical coordinate systems (r, φ , z) is connected with discontinuity at r=0. Vector and scalar Laplace operators contain terms proportional to $\frac{1}{r^2} \frac{\partial}{\partial \varphi}$, which are in general non-integrateable. This discontinuity is imposed mathematically and has to be overcomed with some special means.

Basis functions for velocity in cylindrical coordinates satisfying 2π -periodicity conditions may be defined as

$$\mathbf{v}_{ijk} = \mathbf{v}_{ij}(\mathbf{r}, \mathbf{z}) \exp(i \, \mathbf{k} \boldsymbol{\varphi}) \tag{1}$$

where integer number k changes from $-\infty$ to $+\infty$, and k=0 corresponds to axisymmetric state. Functions v_{ijk} have to satisfy the continuity equation which may be written using (1) as

$$\nabla \cdot \mathbf{v}_{ijk} = \left(\frac{1}{r} \frac{\partial (r \mathbf{u}_{ijk})}{\partial r} + \frac{ik}{r} \mathbf{v}_{ijk} + \frac{\partial (w_{ijk})}{\partial z}\right) \exp(i \, \mathbf{k} \boldsymbol{\varphi}) = 0 \quad (2)$$

where (u, v, w) are r-, φ -, and z- components of v_{iik}

Since three components of the velocity basis are connected with eq. (2) there will be only two independent basis systems in (r, z) plane for |k| > 0, and only one independent basis system for k=0. We define these basis system following [2] as (summation in repeating indices is supposed):

$$\mathbf{v} = \mathbf{A}_{ij}\mathbf{U}_{ij}(\mathbf{r}, \mathbf{z}) + \left(\mathbf{B}_{ijk}\mathbf{V}_{ij}(\mathbf{r}, \mathbf{z}) + \mathbf{C}_{ijk}\mathbf{W}_{ij}(\mathbf{r}, \mathbf{z})\right) \exp(i \, \mathbf{k} \phi) \quad (3)$$

where functions U_{ij} describe axisymmetric part of 3D motion in (r, z)-plane, and functions V_{ij} and W_{ij} describe remaining parts of motion in (r- φ) and (φ -z) planes. A_{ijk} , B_{ijk} , and C_{ijk} are unknown scalar coefficients to be obtained numerically. Functions U_{ij} , V_{ij} and W_{ij} have correspondingly φ -, z-, and r- components equal to zero.

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Using (1) the following formulae for Δv may be evaluated (case $k \neq 0$):

$$\Delta \mathbf{v}_{r} = \left(\frac{\partial^{2} \mathbf{v}_{r}}{\partial r^{2}} + \frac{3}{r} \frac{\partial \mathbf{v}_{r}}{\partial r} + \frac{1 \cdot k^{2}}{r^{2}} \mathbf{v}_{r} + \frac{\partial^{2} \mathbf{v}_{r}}{\partial z^{2}} + \frac{2}{r} \frac{\partial \mathbf{v}_{z}}{\partial z}\right) \exp(i \, \mathbf{k} \phi)$$
$$\Delta \mathbf{v}_{\phi} = \left(\frac{\partial^{2} \mathbf{v}_{\phi}}{\partial r^{2}} + \frac{1}{r} \frac{\partial \mathbf{v}_{\phi}}{\partial r} + \frac{k^{2} \cdot 1}{r^{2}} \mathbf{v}_{\phi} + \frac{\partial^{2} \mathbf{v}_{\phi}}{\partial z^{2}}\right) \exp(i \, \mathbf{k} \phi) - \frac{2i \, \mathbf{k}}{r} \left(\frac{\partial \mathbf{v}_{r}}{\partial r} + \frac{\partial \mathbf{v}_{z}}{\partial z}\right) \exp(i \, \mathbf{k} \phi) - \frac{2i \, \mathbf{k}}{r} \left(\frac{\partial \mathbf{v}_{r}}{\partial r} + \frac{\partial \mathbf{v}_{z}}{\partial z}\right) \exp(i \, \mathbf{k} \phi)$$
(4)
$$\Delta \mathbf{v}_{z} = \left(\frac{\partial^{2} \mathbf{v}_{z}}{\partial r^{2}} + \frac{1}{r} \frac{\partial \mathbf{v}_{z}}{\partial r} - \frac{k^{2}}{r^{2}} \mathbf{v}_{z} + \frac{\partial^{2} \mathbf{v}_{z}}{\partial z^{2}}\right) \exp(i \, \mathbf{k} \phi)$$

As it is seen from (4) in the case of k=1 terms proportional to r^{-2} in expressions for Δv_r and Δv_{φ} may be omitted. So, r- and φ - components of velocity basis corresponding to the first azimuthal mode (k=1) may have non-zero values at the axis. At the same time non-zero value of v_z at the axis must be included in axisymmetric part of the basis coresponding to k=0. These requirements do not influence completeness of the basis, because values of functions at the axis do not depend on the azimuthal coordinate and may be included in any azimuthal mode.

Now we can define polynomial bases for U_{ij} , V_{ij} and W_{ij} for a cylinder of radius l and height L using linear superpositions of Chebyshev polynomials as it is described in [7]:

$$\mathbf{U}_{ij} = \begin{vmatrix} \frac{1}{2} r \sum_{l=0}^{4} a_{il} T_{j+l}(r) & \sum_{l=0}^{4} b_{jl} U_{j+l-l}(\frac{Z}{L}) \\ 0 \\ - \sum_{l=0}^{4} a_{il} \tilde{U}_{j+l-l}(r) & \sum_{l=0}^{4} \frac{b_{jl}}{2(j+l)} T_{j+l}(\frac{Z}{L}) \end{vmatrix}$$
(5)
$$\mathbf{V}_{ij} = \begin{vmatrix} -i \ k \ r^{\alpha} \sum_{l=0}^{4} C_{il} T_{i+l}(r) & \sum_{l=0}^{4} d_{jl} T_{j+l}(\frac{Z}{L}) \\ \sum_{l=0}^{4} C_{il} \tilde{U}_{i+l}(r) & \sum_{l=0}^{4} d_{jl} T_{j+l}(\frac{Z}{L}) \\ 0 \end{vmatrix}$$
(6)

$$\mathbf{W}_{ij} = \begin{bmatrix} \mathbf{v}_{i=0}^{4} \mathbf{e}_{ii} \mathbf{T}_{i+1}(\mathbf{r}) & \sum_{l=0}^{4} \mathbf{f}_{jl} \mathbf{U}_{j+l-1}(\frac{\mathbf{Z}}{L}) \\ -i \ \mathbf{k} \ \mathbf{r} \sum_{l=0}^{4} \mathbf{e}_{il} \mathbf{T}_{i+1}(\mathbf{r}) & \sum_{l=0}^{4} \frac{\mathbf{f}_{jl}}{2(j+1)} \mathbf{T}_{j+1}(\frac{\mathbf{Z}}{L}) \end{bmatrix}$$
(7)

Here $\alpha=0$ for k=1, and $\alpha=1$, for |k|>1,

$$\widetilde{U}_{n}(r) = T_{n+1}(r) + (n+1) r U_{n}(r),
 \widetilde{U}_{n}(r) = (\alpha+1) r^{\alpha} T_{n}(r) + 2n r^{\alpha+1} U_{n-1}(r),$$
(8)

 T_n and U_n are the Chebyshev polynomials of the 1st and 2nd type:

$$T_{n}(x) = \cos[n \arccos(2x-1)],$$

$$U_{n}(x) = \frac{\sin[(n+1) \arccos(2x-1)]}{\sin[\arccos(2x-1)]}$$
(9)

Coefficients a_{il} , b_{jl} , c_{il} , d_{jl} , e_{il} , f_{jl} , are used to satisfy linear homogeneous boundary conditions at r=1, and z=0, L (see ref. [7] for details). Because of connection between Chebyshev polynomials T_n and U_n (d/dx) $T_{n+1}(x)$ = 2(n+1) $U_n(x)$ bases (5–7) satisfy the continuity equation (2).

In the case of non-uniform axisymmetric rotation around the axis a basis for axisymmetric φ -component of velocity must be added to bases (5–7). This component may be approximated in the following way:

$$\begin{aligned} v_{\phi}^{0} &= \Omega(r,z) + r \sum_{i=0}^{N_{r}} \sum_{j=0}^{N_{z}} \Phi_{ij}(t) \left[T_{i}(r) + \alpha_{i} T_{i+1}(r) \right] \times \\ &\times \left[T_{j}(\frac{Z}{L}) + \beta_{1j} T_{j+1}(\frac{Z}{L}) + \beta_{2j} T_{j+2}(\frac{Z}{L}) \right] \end{aligned} \tag{10}$$

And in the case of thermal convection the temperature may be approximated as:

In (10) and (11) functions Φ (r, z) and G (r, z) are used to satisfy non-homogeneous boundary conditions. Coefficients α_i , γ_i , β_{il} and δ_{il} are used to satisfy remaining homogeneous linear boundary conditions for the series in (10) and (11).

Using bases (5–7, 10, 11) all the inner products necessary for realization of the Galerkin method may be calculated analytically. Orthogonal properties of the Fourier exponents used as bases in azimuthal direction will lead to separation of instability problem for each azimuthal number k when linear stability analysis is applied. So, analysis of threshold from axisymmetric to asymmetric state needs $3N_rN_z$ velocity modes and N_rN_z modes for a scalar function for each azimuthal number k, where N_r and N_z are number of functions used in r- and z-directions.

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3. TEST CALCULATIONS

Rayleigh-Benard problem for a cylinder heating from below was used for testing the bases (5–7). Cylinders with aspect ratio A=radius/height equal to 0.5, 1, and 2 were considered. In these cases (see [2]) the most unstable azimuthal mode is known to be k=0 for A=1, and k=1 for A=0.5 and 2. In Table 1 critical Rayleigh numbers Ra_{cr} for all three values of the aspect ratio and k=0, 1, 2, 5, and 10 are reported for truncation increasing from 4×4 to 10×10 modes in (r-z) plane. Bold lines correspond to the most unstable azimuthal mode. As it is seen from the table values of Ra_{cr} show rather rapid convergence. Digits coincide for 8×8 and 10×10 truncations are underlined. Obtained values of Ra_{cr} for the most unstable azimuthal modes are in a good agreements with results

 Table 1-3
 Critical Rayleigh numbers for Rayleigh-Benard instability in a finite cylinder with conductind side wall.

R/L = 0.5	4x4 functions	6x6 functions	8x8 functions	10x10 functions	Result from
k=0	11722.541	11715.605	11715.198	11715.160	
k = 1	8014.8416	8008.5328	8008.3597	8008.3554	8012.
k=2	16780.008	16758.989	16758.189	<u>16758.1</u> 49	
k=5	108255.58	106909.42	106867.62	106866.23	11 IX III
k=10	758739.90	732639.92	731857.89	731751.71	1244

$\mathbf{R}/\mathbf{L} = \mathbf{I}$	4 x 4 functions	6x6 functions	8x8 functions	10x10 functions	Result from [2]	
k = 0	2546.1855	2544.4156	2544.4003	2544.3997	2545	
k=1	2906.1968	2901.5954	2901.5377	<u>2901.53</u> 52		
k=2	3377.5722	3371.0596	3371.0317	<u>3371-03</u> 07		
k=5	10026.115	9924.0082	9921.1643	<u>9921.1</u> 284	pt.	
k=10	54098.874	52251.306	52206.535	<u>5220</u> 1.096		

R/L = 2	4x4 functions	6x6 functions	8x8 functions	10x10 functions	Result from [2]
k=()	1904.8573	1886.2003	1886.0739	<u>1886.07</u> 21	
k = 1	1908.4141	1879.2309	1878.9621	1878.9589	1883
k=2	1904.8643	1895.2507	1895.1368	1895.1328	
k=5	2395.7431	2377.5358	2376.5950	2376.5088	
k=10	5877.3254	5662.7037	5656.7663	<u>5656</u> .1979	

Table 4 Critical Rayleigh numbers and critical frequencies for Rayleigh-Benard instability in a rotating finite cylinder with stress-free horizontal and insulating side boundaries.

Ta	kcr		4 x 4 functions	6x6 functions	8x8 functions	10x10 functions	Analy- tical solution [4, 6]
20	1	Ra _{cr} wcr	1536.01 7.9596	1535.8732 7.5919	1535.8742 7.959201	<u>1535.8742</u> <u>7.959201</u>	1535.87 7.96
40	2	Racr wcr	1990.109 8,6735	1990.2565 9.675701	1990.25647 8.6757057	<u>1990.25647</u> <u>8.67570</u> 66	1990.26 8.68
100	3	Racr wcr	3945.677 12.171	3945.3082 12.21503	3945.3010 12.215057	<u>3945.3010</u> 12.215058	3945.30 12.22
500	4	Racr wcr	18148.0 21.902	17622.83 22.6563	17620.6698 22.68997	<u>17620.48</u> 04 <u>22.69</u> 0305	17620.48 22.69

periodic flows.

obtained in [2].

Only results corresponding to the cases k=0, 1 are compared with other numerical results in tables 1-3. Since velocity bases for k>1 are different from those for k=0, 1, additional testing of the case k>1 is necessary. To testify bases (5-7) for k>1 a problem of Rayleigh-Benard instability in a rotating cylinder was considered. It is known for this problem (see [4-6]) that the azimuthal number K_{cr} corresponding to the most unstable azimuthal mode grows with the growth of the Taylor number Ta, and the oscillatory convective instability of quasi-solid rotating fluid takes place. In the case of stress-free conditions at the top and the bottom of a cylinder the Rayleigh-Benard problem may be solved analytically (see [4, 6]), what provide very good data for testing of a numerical method. Convergence of critical values of the Rayleigh number and frequency of oscillations are reported in the Table 4. As it ts seen from the table all the results show good convergence and are in complete agreement with the analytical solution. With growth of the Taylor number convergence becomes slower, but in the case of Ta=500 the described numerical method still provides at least 4 right digits for 10×10 truncation.

4. CONCLUSIONS

The proposed bases allow to realize the Galerkin spectral method for numerical simulation of 3D hydrodynamical flows in confined circular cylinders. Results obtained for the Rayleigh-Benard problem show that the proposed bases provide good approximation of linear Further calculations have to be carried out for analysis of convergence of the proposed method for investigation of threshold from initially unknown axisymmetric flow to asymmetric one, as well as for calculation of 3D 2π -

terms of the Navier-Stokes and heat transfer equations.

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