

# **Chapter 8 A New Simplified Analytical Internal Mechanics Model for Ship Side Structure Damage**

## **8.1 Introduction**

In chapter 5-7, the ALE FE method is proved to be suitable to predict oil tanker side structure damage response taking account of the fluid-structure interaction of the surrounding water and oil in cargo tank. However, this approach is not design oriented since it requires enormous modeling efforts and computing time. For a large commercial vessel, a few days of CPU time on a workstation are required for simulating a collision or grounding accident. Modeling requires, at least, another one to three months. From the viewpoint of designers, such analysis is far from being feasible. The cost of this analysis is prohibitively high. Proper assessment needs some special techniques, which are not transparent to most naval architects. In this regard, simplified analytical methods are highly preferable.

The purpose of this chapter is to develop a new simplified analytical internal mechanic model for calculating the impact force and the dissipated energy caused by the side structure damage, transverse vibration and global motions of struck ship. The energy induced by the transverse vibration and global motions (including surge, sway, roll, pitch and yaw) are taken into account. Series calculations of ship-ship collisions at different striking velocities are discussed and the critical striking velocity is obtained. Compared with existing results obtained by FEM simulation in time domain, the validation of the applicability of new simplified analytical method is employed. Application of new simplified method to mean oil outflow estimation is performed in section 8.6. Conclusions are collected in section 8.7

## **8.2 Previous Simplified Model for Ship Side Structure Damage**

The simplified model for side structure damage developed by Wang et al (2000) is basically an energy method. The idea is that the work of an external force must equal the plastic deformed energy dissipation in the structure. The progressive failure model of a double hull structure is can be seen in Fig. 8.1. Damage of the side structure is a progressive process. With indentation increases further and further, more and more structures become involved. The resistance load increases because of the large volume of

damaged structures. The primary structural failure modes are identified and idealized as penetration of plate, denting of web and tearing of plate.

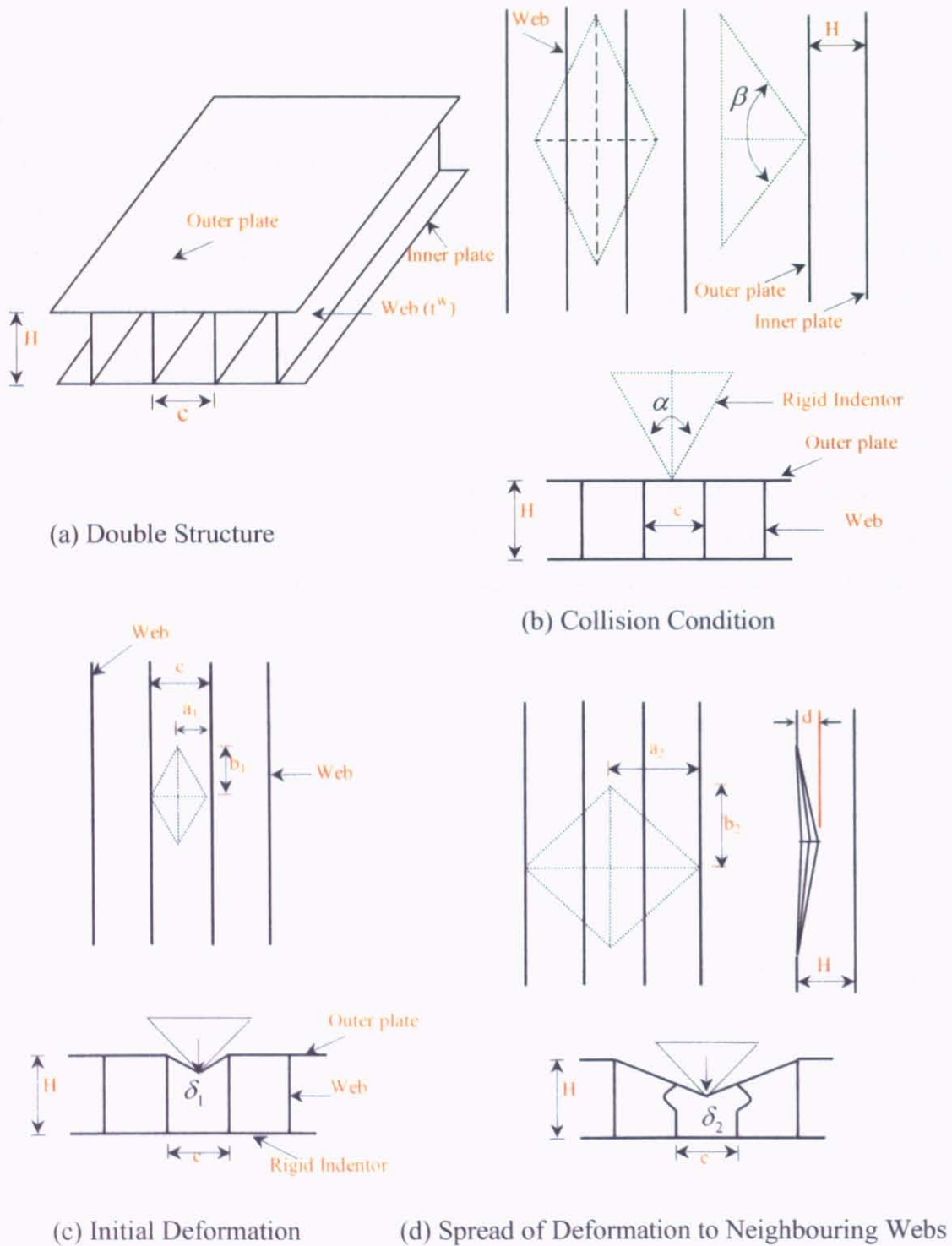


Fig. 8.1 Progressive failure model of a double hull structure

Because concentrated load is usually assumed in researches in side collision situation, the load-carrying capacity of penetration of the side shell is discussed. The rigid bow forces the plasticity spread the whole extent of the plate. After wrapped round the bow, this part of material is assumed to remain rigid. At a given finite deflection, the material within the boundary and wrapped part of plate are assumed to be yielding uniformly. By application of theoretical plasticity methods and minimization of the energy dissipation, the load-carrying capacity of penetration of the side shell is given by:

$$F_p = 2N_0 \frac{a_1^2 + b_1^2}{a_1 b_1} \delta_1 \quad N_0 = \sigma_0 t \quad (8.1)$$

Where  $\sigma_0$  is material flow stress,  $t$  is thickness of shell side plate

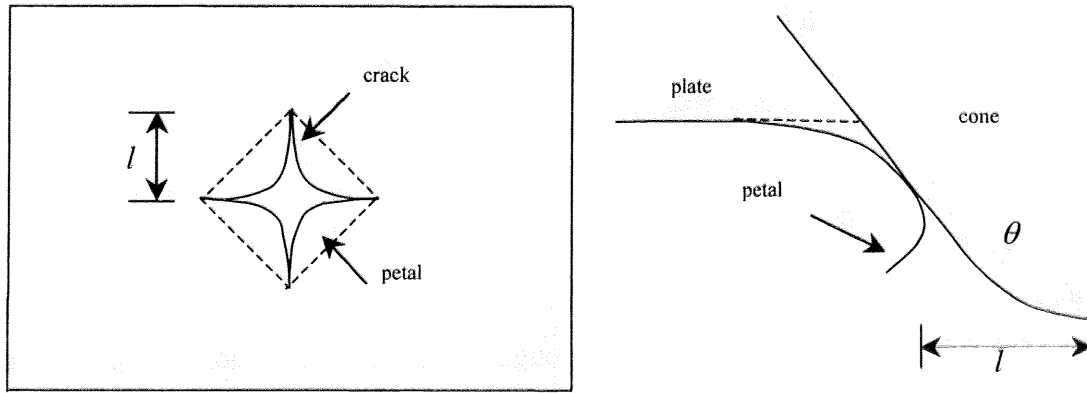


Fig. 8.2 Model for ruptured plate

The plate crack process, as shown in Fig. 8.2, is similar to the plate tearing mechanism. This process involves very large stretching in the vicinity of the striker (bow tip in plate tearing), prominent bending of plate “flaps” or petals remote from striker, and friction. The plate tearing process has been intensively discussed as one of the primary energy absorbing mechanisms in a collision or grounding event. The force needed to drive the crack to a length of  $l$  is given by:

$$F_t = 1.51\sigma_0 t^{1.5} l^{0.5} (\sin \theta)^{0.5} (l + \mu / \tan \theta) \quad (8.2)$$

Where  $l$  is tearing length,  $\theta$  is semi-angle of the wedge and  $\mu$  is friction coefficient.

For a general case, there are  $n$  cracks in a perforated plate. It is observed that the petals bend along the lines connecting the crack tips. The perforation process involves

plastic flow near the crack tips, bending of the petals and friction between the cone and the plate. Two adjacent petals form an angle of  $(n-2)\pi/n$ . When neglecting friction, the force needed to drive one crack to a length of  $l$  is given by:

$$P_n = 1.51\sigma_0 t^{1.5} l^{0.5} (\sin((n-2)\pi/2n))^{0.5} \quad (8.3)$$

Where  $n$  is the number of cracks in ruptured plate,  $P_n$  is in the original plane of the yet not deformed plate.

In order to push the bow into this perforated plate, additional load is required to resist friction. Taking friction into account, the force needed to drive the bow into the perforated plate is given by:

$$F_l = nP_n (\tan \theta + \mu) = 1.51\sigma_0 t^{1.5} l^{0.5} n (\sin((n-2)\pi/2n))^{0.5} (\tan \theta + \mu) \quad (8.4)$$

For denting of webs, the initial phase (before occurrence of buckling) is very short, and the deformation is in the order of plate thickness. Out-of-plane deformation of the web becomes larger and larger as the compression proceeds. If the collision energy is very large, this bulge will eventually evolve into structural folding (see Fig. 8.1). Both bending stress and membrane stress play a role in the energy absorption. If the energy is still not spent, then a new bulge will occur directly behind the fold, and it will develop in a process similar to that of the first-fold. Because web has enough depth in the direction of collision load (compared to the folding length), more than one-fold can be formed. The load-carrying capacity of denting of the web is given by:

$$F_d = 2.32\sigma_0 (2b_1)^{0.33} (t)^{1.67} \quad (8.5)$$

Where  $2b_1$  is width of the denting web.

Strain of side shell plate is

$$\varepsilon = \frac{1}{2} \left( \frac{\delta_1}{a_1} \right)^{1/2} \quad (8.6)$$

In simplified analytical model, the failure criterion is that rupture occurs when the strain equals a prescribed critical rupture stain. Here, it is assumed that a plate will rupture if the strain equals 30%.

The plastic deformed energy and friction energy  $E_{LP}$  dissipation in the side structure can be expressed as

$$E_{LP} = \int_0^{\delta_p} F_C(\delta) \cdot d\delta \quad (8.7)$$

Here, simplified analytical method is used for the side strength analysis of a 293,000 tonne VLCC, which has a collision barrier side structure, against a collision from 350,000 tonne VLCC. The FEM simulations of the collisions case have been carried out in chapter 6. The stiffened side panel is viewed as a plate structure with equivalent thickness. The total resistance of side structure is obtained by summing all the contributions from plate components that are directly indented by the penetrating bow.

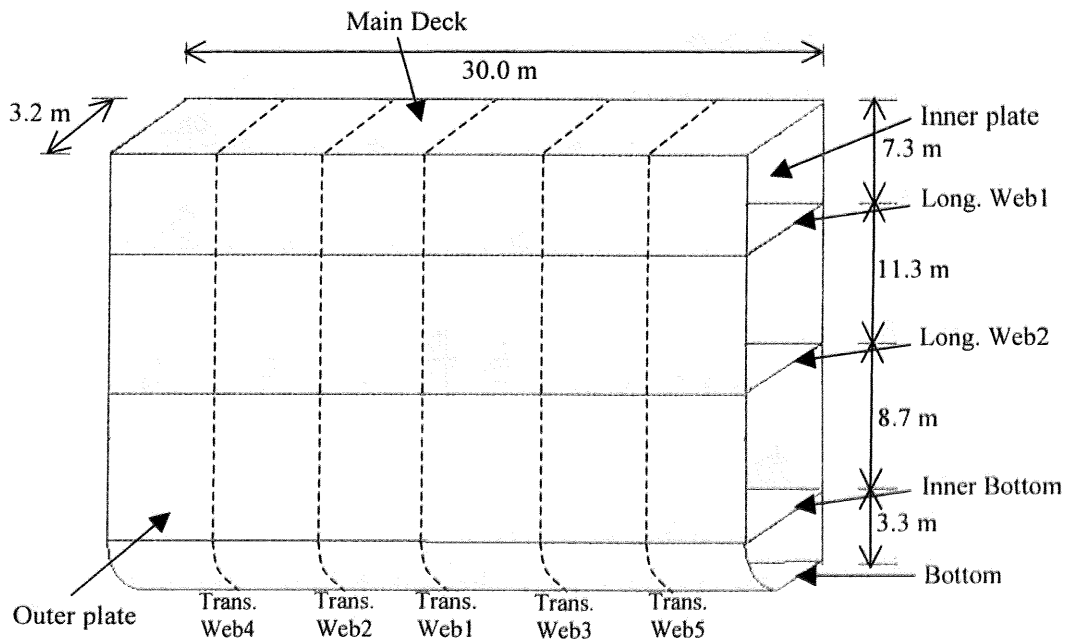


Fig. 8.3 Side structure of struck ship (equivalent thickness is shown)

Brief descriptions of simplified analytical method for side structure of 293,000 tonne VLCC are in the following.

Stage 1 (till the bulb touches inner bottom)

- $\delta = (0 \sim 956) \text{ mm}$
- Penetration of outer plate by bulb, ( $F_{pb}$ )
- Denting of transverse web 1, ( $F_{dT1}$ )
- Total Force,  $F_C(\delta) = F_{pb} + F_{dT1}$

Stage 2 (till the bulb touches transverse web 2, 3)

- $\delta = (956 \sim 1409) \text{ mm}$
- Penetration of outer plate by bulb, ( $F_{pb}$ )
- Denting of transverse web 1, ( $F_{dT1}$ )
- Denting of inner bottom, ( $F_{dIB}$ )
- Total Force,  $F_C(\delta) = F_{pb} + F_{dT1} + F_{dIB}$

Stage 3 (till the bulb tearing outer plate)

- $\delta = (1409 \sim 1800) \text{ mm}$
- Penetration of outer plate by bulb, ( $F_{pb}$ )
- Denting of transverse web 1, 2, 3, ( $F_{dT1}, F_{dT2}, F_{dT3}$ )
- Denting of inner bottom, ( $F_{dIB}$ )
- Total Force,  $F_C(\delta) = F_{pb} + F_{dT1} + F_{dT2} + F_{dT3} + F_{dIB}$

Stage 4 (till the stem touches outer plate and main deck)

- $\delta = (1800 \sim 1958) \text{ mm}$
- Tearing of outer plate by bulb, ( $F_{tb}$ )
- Denting of transverse web 1, 2, 3, ( $F_{dT1}, F_{dT2}, F_{dT3}$ )
- Denting of inner bottom, ( $F_{dIB}$ )
- Total Force,  $F_C(\delta) = F_{tb} + F_{dT1} + F_{dT2} + F_{dT3} + F_{dIB}$

Stage 5 (till the bulb touches inner plate)

- $\delta = (1958 \sim 3196) \text{ mm}$
- Tearing of outer plate by bulb, ( $F_{tb}$ )
- Penetration of outer plate by stem, ( $F_{ps}$ )
- Denting of main deck by stem, ( $F_{dm}$ )
- Denting of transverse web 1, 2, 3, ( $F_{dT1}, F_{dT2}, F_{dT3}$ )
- Denting of longitudinal web 1 and inner bottom, ( $F_{dL1}, F_{dIB}$ )
- Total Force,  $F_C(\delta) = F_{tb} + F_{ps} + F_{dm} + F_{dT1} + F_{dT2} + F_{dT3} + F_{dL1} + F_{dIB}$

Stage 6 (till the stem tearing of outer plate)

- $\delta = (3196 \sim 3793) \text{ mm}$
- Tearing of outer plate by bulb, ( $F_{tb}$ )
- Penetration of inner plate by bulb, ( $F_{pb}$ )
- Penetration of outer plate by stem, ( $F_{ps}$ )
- Denting of main deck by stem, ( $F_{dm}$ )
- Denting of transverse web 1, 2, 3, ( $F_{dT1}, F_{dT2}, F_{dT3}$ )
- Denting of longitudinal web 1 and inner bottom, ( $F_{dL1}, F_{dIB}$ )
- Total Force,

$$F_C(\delta) = F_{tb} + F_{pb} + F_{ps} + F_{dm} + F_{dT1} + F_{dT2} + F_{dT3} + F_{dL1} + F_{dLB}$$

Stage 7 (till the bulb tearing of inner plate)

- $\delta = (3793 \sim 7307)$  mm
- Tearing of outer plate by bulb, ( $F_{tb}$ )
- Penetration of inner plate by bulb, ( $F_{pb}$ )
- Tearing of outer plate by stem, ( $F_{ts}$ )
- Denting of main deck by stem, ( $F_{dm}$ )
- Denting of transverse web 1, 2, 3, ( $F_{dT1}, F_{dT2}, F_{dT3}$ )
- Denting of longitudinal web 1, 2 and inner bottom, ( $F_{dL1}, F_{dL2}, F_{dLB}$ )
- Total Force,

$$F_C(\delta) = F_{tb} + F_{pb} + F_{ts} + F_{dm} + F_{dT1} + F_{dT2} + F_{dT3} + F_{dL1} + F_{dL2} + F_{dLB}$$

Table 8.1 Force and Energy absorbed at each stage of penetration

Stage	Penetration (mm)	Force(KN)
1	956	2606
2	1409	4085
3	1800	5887
4	1958	18374
5	3196	53662
6	3793	82155
7	7307	223060

The impact force-penetration curve for side structure of 293,000 tonne VLCC till the bulbous bow tearing of inner shell is shown in Fig.8.4. Compared with FEM simulation result, impact force-penetration curve using previous simplified method is underestimated. The main reason is that the overall elastic transverse vibration response and the whole kinetic energy of struck ship (including surge, sway, roll, yaw and pitch motion) are neglected. ALE FEM simulation results of energy components in Fig.8.5 show that the elastic bending energy caused by transverse vibration and global kinetic energy can not be neglected. In the next section, the new simplified analytical model, which is considered the effect of these factors, is proposed.

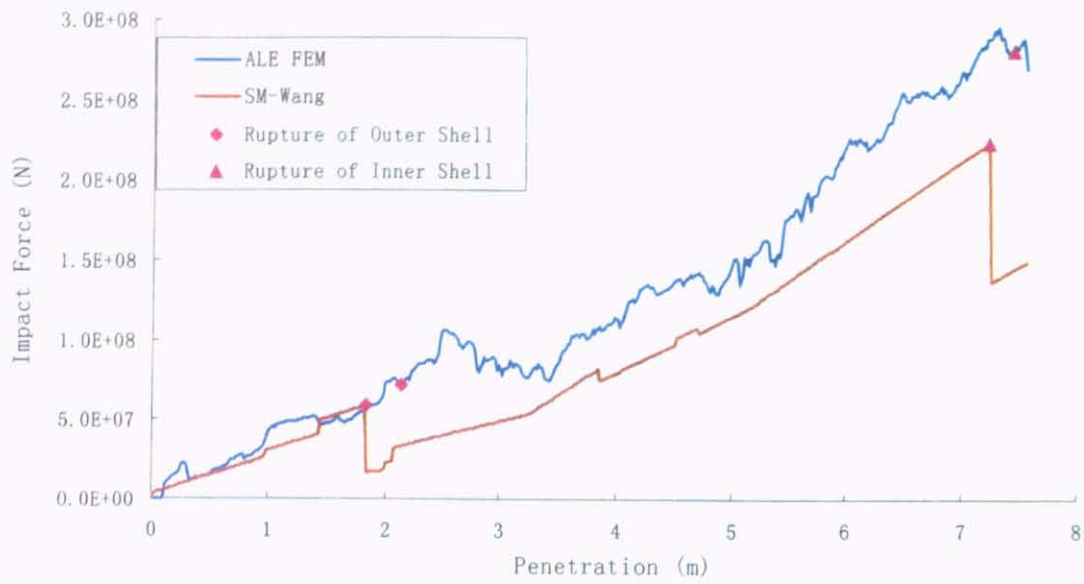


Fig. 8.4 Impact force-penetration curve

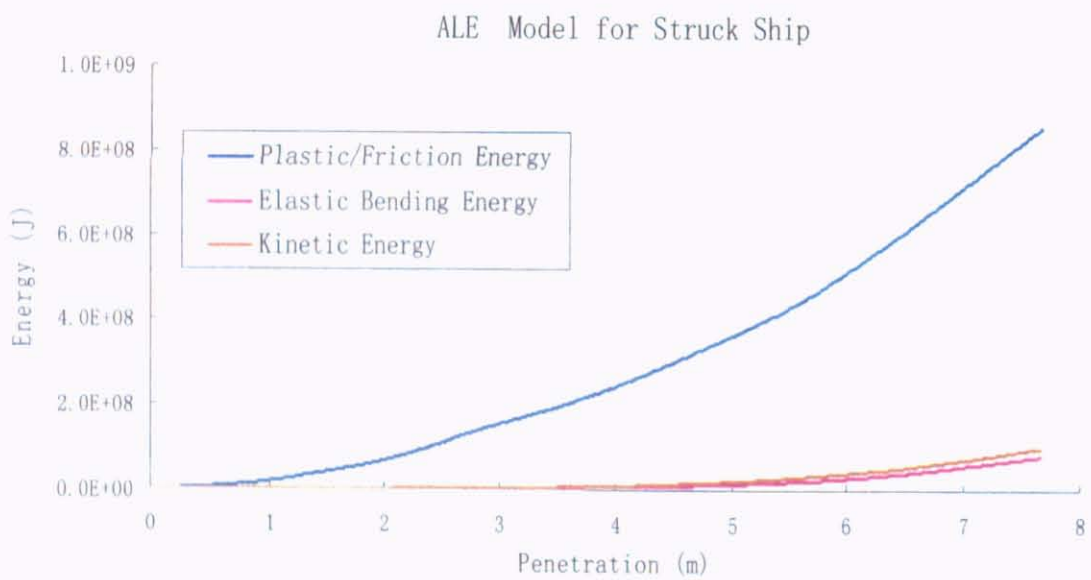


Fig. 8.5 Energy components of FEM simulation result

### 8.3 New Simplified Analytical Model for Side Structure Damage

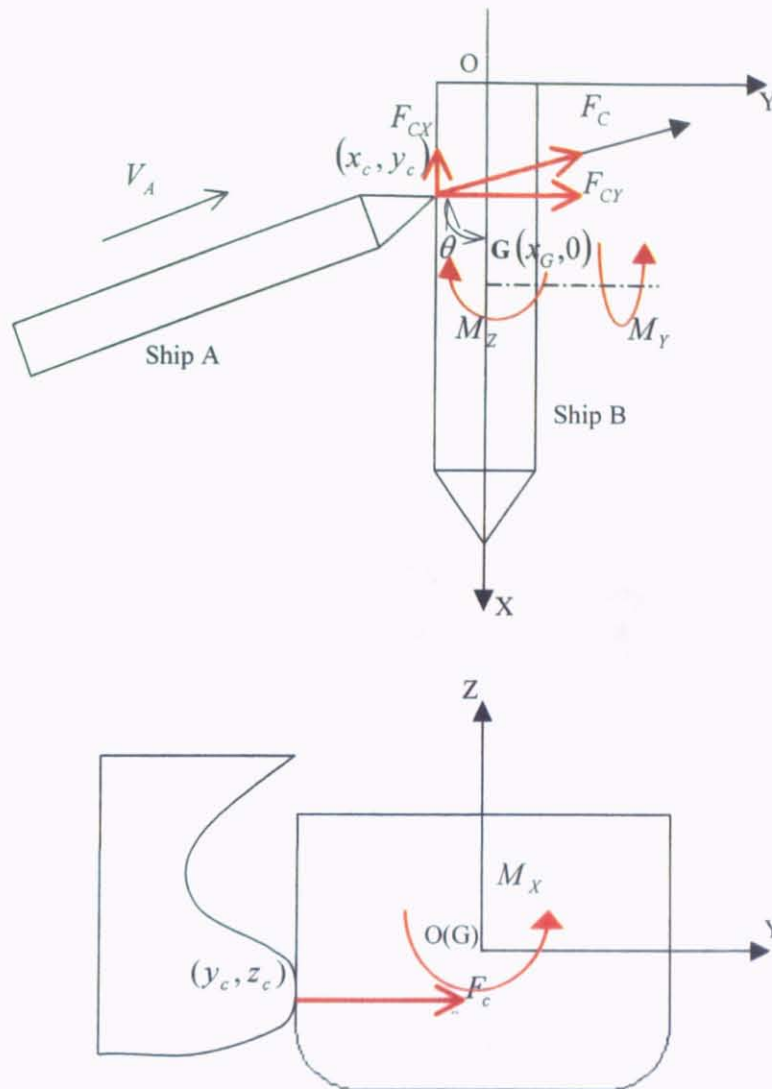


Fig. 8.6 Plan view of arbitrary ship-ship collision

In this section, a new simplified analytical internal mechanics model, which the effects of the overall elastic transverse vibration response, the whole kinetic energy of struck ship(including surge, sway, roll, yaw and pitch motion) on the impact force are also considered, is proposed. As Fig.8.6 shown, the rigid striking ship collides the side structure of struck ship at arbitrary collision location and angle, the impact force will cause the transverse vibration, the roll, yaw, pitch, sway and surge motion of struck ship. For simple, we assume that the striking ship is rigid and the struck ship is standstill.

### 8.3.1 The Simplified Mechanics Model for Roll Motion of Struck Ship

In section 6.3, ALE FE simulation results show that the roll motion of struck ship is significantly affected the side structure damage process. Endo et al (2004) proposed a simplified spring model to simulate the heave and roll motion in Fig. 8.7.

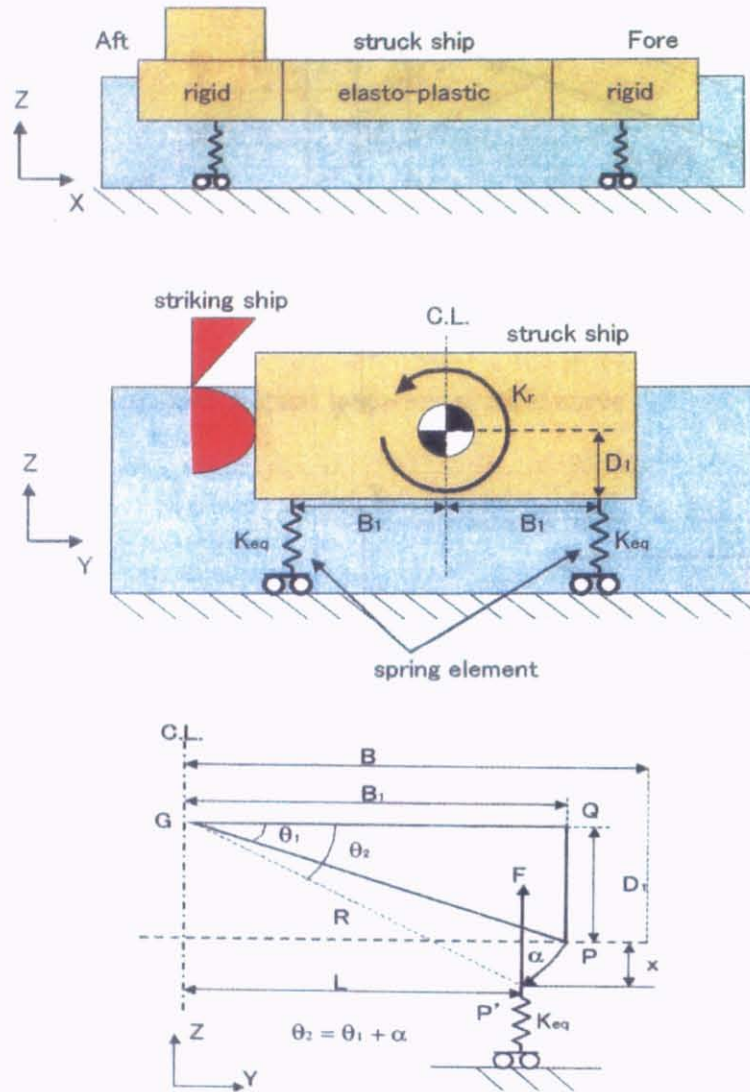


Fig. 8.7 Simplified Spring Model for Roll Motion

The stiffness of 4 spring elements is determined by static restoring moment for the heave motion

$$K_{eq} = \frac{K_h}{4} \quad (8.8)$$

Where  $K_h$  is the restoring moment coefficient for heave motion.

The restoring moment using spring model for roll motion is

$$M_{spring} = 4K_{eq} \cdot x \cdot R \cdot \cos(\theta_2) = 4K_{eq} \cdot R^2 \cdot (\sin \theta_2 - \sin \theta_1) \cos \theta_2 \quad (8.9)$$

$$\tan \theta_1 = \frac{D_1}{B_1} \quad R^2 = D_1^2 + B_1^2 \quad \theta_2 = \theta_1 + \alpha \quad (8.10)$$

The restoring moment can also be expressed as

$$M_w = K_r \cdot \alpha \quad (8.11)$$

Where  $\alpha$  is the roll angle,  $B_1$  is the distance from centre line to spring element,  $D_1$  is the vertical distance from bottom to center of gravity.

Assuming  $\alpha$  is small, then  $\cos \alpha \approx 1$ ,  $\sin \alpha \approx \alpha$ ,  $\alpha^2 \approx 0$

$$\begin{aligned} \sin \theta_2 &= \sin(\theta_1 + \alpha) = \sin \theta_1 \cos \alpha + \cos \theta_1 \sin \alpha \approx \sin \theta_1 + \alpha \cos \theta_1 \\ \cos \theta_2 &= \cos(\theta_1 + \alpha) = \cos \theta_1 \cos \alpha - \sin \theta_1 \sin \alpha \approx \cos \theta_1 - \alpha \sin \theta_1 \end{aligned} \quad (8.12)$$

The restoring moment using spring model for roll moment can be expressed as

$$M_{spring} = 4K_{eq} \cdot \alpha \cdot R^2 \cdot (\cos \theta_1)^2 = 4K_{eq} \cdot \alpha \cdot B_1^2 \quad (8.13)$$

Finally, we can determine the value of  $B_1$

$$B_1 = \frac{1}{2} \sqrt{\frac{K_r}{K_{eq}}} \quad (8.14)$$

The potential energy of spring can be expressed as

$$E_{Roll} = 4 \times \frac{1}{2} K_{eq} \cdot \Delta x^2 = 2 \times K_{eq} \cdot (D_1^2 + B_1^2) \left[ \sin \left( \theta_1 + \frac{F_C \sin \theta \cdot h}{K_r} \right) - \sin \theta_1 \right]^2 \quad (8.15)$$

Where  $h$  is the vertical distance between the contact point and the centre of gravity.

Here, this spring model is used in FE model to analyze the 293,000 tonne VLCC collided by the 350,000 VLCC (the same ship collision case in section 6.4). The impact force-penetration curve of roll-spring model is presented in Fig. 8.7. Compared with ALE FE model result, the roll-spring model can give good prediction results on the rupture of outside shell plate and inner plate. The spring model can consider the effect of roll motion on the structure damage well.

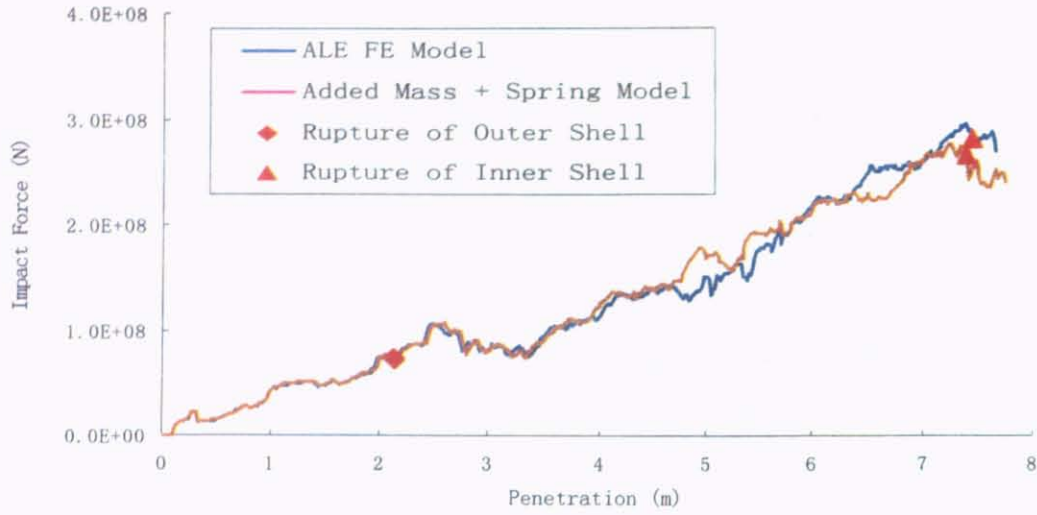


Fig. 8.8 Impact force-penetration curve for roll-spring model

### 8.3.2 Kinetic Energy of Struck Ship for Surge, Sway, Pitch and Yaw Motion

#### Motion equations

Here, the transverse displacements of struck ship for surge and sway are denoted by  $u^b, v^b$  respectively. The angular displacements of the rotational motion for pitch and yaw are denoted by  $\varphi^b, \psi^b$  respectively.

The motions of struck ship for surge, sway, pitch and yaw can be expressed respectively

$$(M^b + A_{1,1}^b) \ddot{u}^b = -F_C \cos \theta \quad (8.16)$$

$$(M^b + A_{2,2}^b) \ddot{v}^b = F_C \sin \theta \quad (8.17)$$

$$(I_{yy}^b + A_{5,5}^b) \ddot{\varphi}^b = -F_C \cos \theta \cdot z_c \quad (8.18)$$

$$(I_{zz}^b + A_{6,6}^b) \ddot{\psi}^b = -F_C \sin \theta \cdot (x_G - x_c) - F_C \cos \theta \cdot \frac{B}{2} \quad (8.19)$$

where  $M^b, A_{1,1}^b$  and  $A_{2,2}^b$  present mass and added mass for surge, sway motion of struck ship;  $I_{yy}^b$  and  $I_{zz}^b$  present the moment of inertia with respect to y-axis and z-axis;  $A_{5,5}^b$  and  $A_{6,6}^b$  present added mass moment of inertia for pitch and yaw.

## Calculation of added mass coefficients

For simplicity, the calculation of added mass coefficients in motion equations is given as following.

The hydrodynamic force related to the surge motion of ship is very small compared to sway force. The hydrodynamic forces related to the surge motion cannot be found by strip method. The sectional surge added mass  $A_{1,1}$  related to the forward motion is small compared with the mass of the ship. Professor Motora (1969) found it to be  $(0.02 - 0.07)M$ . A reasonable assumption may be

$$A_{1,1} = 0.05M \quad (8.20)$$

Many researchers have proposed different methods to evaluate the sectional sway added mass coefficients for ship forms; the pioneer works being that by Lewis F. M. (1929), Tasai F. (1961). The sway hydrodynamic added mass coefficient at infinite frequency using Lewis Two Parameter Conformal Mapping Method to be given by

$$A_{2,2} / M = \frac{2\rho}{\pi} \left\{ D^2 + \frac{16}{3} a_3^3 \right\} \quad (8.21)$$

where

$$a_3 = \frac{-c_1 + 3 + (9 - 2c_1)^{1/2}}{c_1} \quad (8.22)$$

$$c_1 = \left[ 3 + \frac{4\sigma_s}{\pi} \right] + \left[ 1 - \frac{4\sigma_s}{\pi} \right] [(H_o - 1)/(H_o + 1)]^2 \quad (8.23)$$

$$H_o = \frac{B}{2D} \quad \sigma_s = \frac{A_s}{DB} \quad (8.24)$$

$B$  and  $D$  is the breadth and draught of cross section, respectively

$A_s$  is wetted area of cross section

The added mass coefficient for the pitch/yaw motion of ship,  $A_{5,5} / A_{6,6}$ , is (Pedersen et al., 1993)

$$A_{5,5} = 0.21 \cdot I_{yy} \quad (8.25)$$

$$A_{6,6} = 0.21 \cdot I_{zz} \quad (8.26)$$

## Kinetic energy for different motions

We assume the external force to a stepwise one of magnitude  $F_c$  for a duration  $\delta t$ . Then the velocity/angle velocity of struck ship for surge, sway, pitch and yaw can be expressed respectively

$$\dot{u}^b = \int \ddot{u}^b dt = \int \frac{-F_c \cos \theta}{M^b + A_{1,1}^b} dt = \frac{-F_c \cos \theta \cdot \delta t}{M^b + A_{1,1}^b} + C_1 \quad (8.27)$$

$$\dot{v}^b = \int \ddot{v}^b dt = \int \frac{F_c \sin \theta}{M^b + A_{2,2}^b} dt = \frac{F_c \sin \theta \cdot \delta t}{M^b + A_{2,2}^b} + C_2 \quad (8.28)$$

$$\dot{\phi}^b = \int \ddot{\phi}^b dt = \int \frac{-F_c \cos \theta \cdot z_c}{I_{yy}^b + A_{5,5}^b} dt = \frac{-F_c \cos \theta \cdot z_c \cdot \delta t}{I_{yy}^b + A_{5,5}^b} + C_3 \quad (8.29)$$

$$\dot{\psi}^b = \int \ddot{\psi}^b dt = \frac{\left( -F_c \sin \theta \cdot (x_G - x_c) - F_c \cos \theta \cdot \frac{B}{2} \right) \cdot \delta t}{I_{zz}^b + A_{6,6}^b} + C_4 \quad (8.30)$$

At  $t = 0$ , the constant  $C_1, C_2, C_3$  and  $C_4$  become the initial velocity of struck ship for surge, sway, pitch and yaw motion. During the next time step,  $C_1, C_2, C_3$  and  $C_4$  take the value of  $\dot{u}^b, \dot{v}^b, \dot{\phi}^b$  and  $\dot{\psi}^b$  at the end of the previous time step).

The kinetic energy of struck ship for surge motion can be expressed as

$$E_{K\text{Surge}} = \frac{1}{2} (M^b + A_{1,1}^b) \cdot (\dot{u}^b)^2 = \frac{(M^b + A_{1,1}^b)}{2} \cdot \left( \frac{-F_c \cos \theta \cdot \delta t}{M^b + A_{1,1}^b} + C_1 \right)^2 \quad (8.31)$$

The kinetic energy of struck ship for sway motion can be expressed as

$$E_{K\text{Sway}} = \frac{1}{2} (M^b + A_{2,2}^b) \cdot (\dot{v}^b)^2 = \frac{(M^b + A_{2,2}^b)}{2} \cdot \left( \frac{F_c \sin \theta \cdot \delta t}{M^b + A_{2,2}^b} + C_2 \right)^2 \quad (8.32)$$

The kinetic energy of struck ship for pitch motion can be expressed as

$$E_{K\text{Pitch}} = \frac{1}{2} (I_{yy}^b + A_{5,5}^b) \cdot (\dot{\phi}^b)^2 = \frac{(I_{yy}^b + A_{5,5}^b)}{2} \cdot \left( \frac{-F_c \cos \theta \cdot z_c \cdot \delta t}{I_{yy}^b + A_{5,5}^b} + C_3 \right)^2 \quad (8.33)$$

The kinetic energy of struck ship for yaw motion can be expressed as

$$E_{KYaw} = \frac{1}{2} (I_{zz}^b + A_{6,6}^b) \cdot (\dot{\psi}^b)^2 = \frac{(I_{zz}^b + A_{6,6}^b)}{2} \cdot \left( \frac{\left( -F_C \sin \theta \cdot (x_G - x_c) - F_C \cos \theta \cdot \frac{B}{2} \right) \cdot \delta}{I_{zz}^b + A_{6,6}^b} + C_4 \right)^2 \quad (8.34)$$

Total kinetic energy of struck ship for surge, sway, roll, pitch and yaw motion can be expressed as

$$E_{GK} = E_{KSurge} + E_{KSway} + E_{KPitch} + E_{KYaw} + E_{Roll} \quad (8.35)$$

### 8.3.3 The Hull Transverse Vibratory Response

Impact loading on a ship induces not only the rigid body motions, but also the dynamic bending of the ship hull girder. Dynamic bending covers the hull girder vibration where the cross-sections of the beam remain plane. This allows the modelling of the struck ship hull vibratory as an Euler-Bernoulli beam with free ends and the bending vibration is considered in the plane of the water surface. The transverse hull forced vibration of struck ship can be expressed:

$$EI \frac{\partial^4 v^b(x,t)}{\partial x^4} + C \frac{\partial v^b(x,t)}{\partial t} + \rho \frac{\partial^2 v^b(x,t)}{\partial t^2} = F_C \sin \theta \cdot \delta(x_c) \quad (8.36)$$

$$\rho = \frac{(M^b + A_{2,2}^b)}{L} \quad (8.37)$$

where  $M^b$  and  $A_{2,2}^b$  present mass and added mass for sway motion of struck ship;  $L$  and  $EI$  are the length and flexural stiffness of hull beam; the viscous damping coefficient is denoted  $C$ ;  $\theta$  is collision angle and  $\delta(x_c)$  is 1 when  $x = x_c$  and 0 for other value of  $x$ .

The dynamic response of the ship hull girder is a superposition of the responses of the different eigenmodes. The essential operation of the mode-superposition analysis is the transformation from the geometric displacement coordinates to the normal coordinates. The transverse displacement is expressed as

$$v^b(x,t) = \phi(x) \cdot Y(t) \quad (8.38)$$

Where  $\phi(x)$  is the natural mode and  $Y(t)$  is the normal coordinate

Inserting equation (8.38) into equation (8.36), the equation of  $\phi(x)$  can be expressed as

$$EI \frac{d^4 \phi(x)}{dx^4} - \rho \omega^2 \cdot \phi(x) = 0 \quad (8.39)$$

$$\beta^4 = \omega^2 \cdot \frac{\rho}{EI} \quad (8.40)$$

The solution of natural mode  $\phi(x)$  is

$$\phi(x) = A_1 \sin \beta x + A_2 \cos \beta x + A_3 \sinh \beta x + A_4 \cosh \beta x \quad (8.41)$$

For the boundary condition of free-free-end hull beam vibration

$$\begin{cases} \phi''(0) = \phi'''(0) = 0 \\ \phi''(L) = \phi'''(L) = 0 \end{cases} \quad (8.42)$$

We can get

$$\begin{cases} A_1 = A_3 \\ A_2 = A_4 \\ \cos \beta L \cdot \cosh \beta L = 1 \end{cases} \quad (8.43)$$

$$\beta_k L \approx \left(k + \frac{1}{2}\right)\pi + (-1)^{k+1} \cdot 2e^{-\left(k+\frac{1}{2}\right)\pi} \quad k \geq 1 \quad (8.44)$$

For  $k$ -th natural frequency

$$\omega_k = \beta_k^2 \cdot \sqrt{\frac{EI}{\rho}} \quad (8.45)$$

The normal natural mode

$$\phi_n(x) = A_n \left( \sin \beta_n x + \sinh \beta_n x - \frac{\sin \beta_n L - \sinh \beta_n L}{\cos \beta_n L - \cosh \beta_n L} (\cos \beta_n x + \cosh \beta_n x) \right) \quad (8.46)$$

Furthermore the effect of vibration motion is assumed to be small compared to the rigid body motion, only the first eigenmode is include.

$$\omega_1 = \left(\frac{4.73}{L}\right)^2 \sqrt{\frac{EI}{\rho}} \quad (8.47)$$

$$\phi_1(x) = A_1 \cdot \left( \sin \frac{4.73}{L} x - 1.02 \cos \frac{4.73}{L} x + \sinh \frac{4.73}{L} x - 1.02 \cosh \frac{4.73}{L} x \right) \quad (8.48)$$

Since  $\phi_1(x)$  satisfies the normality condition

$$\int_0^L \rho \phi_1^2(x) dx = 1 \quad A_1 = 0.97 \sqrt{\frac{1}{\rho L}} \quad (8.49)$$

Then  $\phi_1(x)$  can be expressed as

$$\phi_1(x) = \frac{0.97}{\sqrt{\rho L}} \cdot \left( \sin \frac{4.73}{L} x - 1.02 \cos \frac{4.73}{L} x + \sinh \frac{4.73}{L} x - 1.02 \cosh \frac{4.73}{L} x \right) \quad (8.50)$$

Introducing equation (8.38) into (8.36), multiplying the result by  $\phi_1(x)$ , integrating over the domain  $0 < x < L$

$$Y_1(t) \int_0^L EI \frac{d^4 \phi_1(x,t)}{dx^4} \phi_1(x) dx + C \dot{Y}_1(t) \int_0^L \phi_1^2(x) dx + \ddot{Y}_1(t) \int_0^L \rho \phi_1^2(x) dx = \int_0^L \phi_1(x) F_C dx \quad (8.51)$$

Considering the following condition

$$\int_0^L EI \frac{d^4 \phi_1(x,t)}{dx^4} \phi_1(x) dx = \omega_1^2 \quad (8.52)$$

Equation of  $Y_1(t)$  can be expressed as

$$\ddot{Y}_1(t) + 2\xi_1 \omega_1 \dot{Y}_1(t) + \omega_1^2 Y_1(t) = P_1 \quad (8.53)$$

$$P_1 = \int_0^L \phi_1(x) \cdot F_C \sin \theta \delta(x_c) dx = \phi_1(x) \cdot F_C \sin \theta \Big|_{x=x_c} \quad (8.54)$$

$$\bar{\omega}_1 = \omega_1 \cdot \sqrt{1 - \xi^2} \quad (8.55)$$

The solution to  $Y_1(t)$  is given by:

$$Y_1(t) = e^{-\xi_1 \omega_1 t} (B_1(t) \cos \bar{\omega}_1 t + C_1(t) \sin \bar{\omega}_1 t) + D_1(t) \quad (8.56)$$

where

$$B_1(t) = \rho \int_0^L v_1(x,0) \cdot \phi_1(x) dx \quad (8.57)$$

$$C_1(t) = \frac{\rho}{\bar{\omega}_1} \int_0^L (\dot{v}_1(x,0) + \xi_1 \omega_1 v_1(x,0)) \dot{v}_1(x,0) \cdot \phi_1(x) dx \quad (8.58)$$

$$D_1(t) = \frac{1}{\bar{\omega}_1} \int_0^t P_1 \cdot e^{-\xi_1 \omega_1 (t-\tau)} \sin[\bar{\omega}_1 (t-\tau)] d\tau \quad (8.59)$$

The value of damping factor  $\xi$  is usually obtained experimentally. If no empirical value exist for a particular ship, measured internal damping values from many ships are

presented in reference (ISSC, 1983). These values indicate that damping factor is practically independent of frequency and a value  $\xi = 0.05$  may be used.

Bending moment induced by the transverse hull beam vibration is

$$M(x, t) = EI \cdot \frac{\partial^2 v(x, t)}{\partial^2 x} \quad (8.60)$$

The global elastic bending energy induced by transverse vibration

$$E_{vB}(x, t) = \int_0^L \frac{M^2(x, t)}{2EI} dx = \int_0^L \frac{EI}{2} \cdot \left( \frac{\partial^2 v(x, t)}{\partial x^2} \right)^2 dx \quad (8.61)$$

The kinetic energy of hull beam transverse vibration can be expressed

$$E_{vK}(x, t) = \frac{1}{2} \int_0^L \rho \cdot \left( \frac{\partial v(x, t)}{\partial t} \right)^2 dt \quad (8.62)$$

The total energy caused by transverse vibration is

$$E_v(x, t) = E_{vB} + E_{vK} \quad (8.63)$$

### 8.3.4 New Simplified Analytical Internal Mechanics Model

Here, a new simplified analytical internal mechanics model is proposed. The idea is that the work of an external force must equal the plastic deformed energy dissipation in the local structure  $E_{LP}$ , elastic bending energy and kinetic induced by transverse hull vibration  $E_v$ , kinetic energy of struck ship for surge, sway, roll, pitch and yaw motion  $E_{GK}$ .

The new simplified analytical internal mechanics model can be expressed as

$$\int_0^{\delta_n} F_C^{New}(\delta) \cdot d\delta = \int_0^{\delta_n} (F^{prev}(\delta) + F_{Dyna}(\delta)) \cdot d\delta = E_{LP} + E_v + E_{GK} \quad (8.64)$$

## 8.4 Verification of New Simplified Analytical Internal Mechanics Model

In order to check the accuracy of the present simplified analytical internal mechanics model, the same ship collision case in section 6.4 is used. The relative parameters for this side ship collision case are as following:

Initial velocity of striking ship:

$$V_0^a = 16 \text{ knot}$$

Collision location:

$$x_c = \frac{L}{2} = 163.5 \text{ m (midship)}$$

Collision angle:

$$\theta = 90^\circ$$

Flexural stiffness of struck ship hull beam:

$$EI = 6.3 \times 10^{14} \text{ N} \cdot \text{m}^2$$

Damping factor:

$$\zeta = 0.05$$

Added mass coefficients:

$$\frac{A_{11}^b}{M^b} = 0.05; \quad \frac{A_{22}^b}{M^b} = 0.24;$$

$$\frac{A_{55}^b}{I_{yy}^b} = \frac{A_{66}^b}{I_{zz}^b} = 0.21$$

Spring model parameters for roll motion:

$$K_{eq} = 3.66 \times 10^7 \text{ N/m}$$

$$B_1 = 11.46 \text{ m}$$

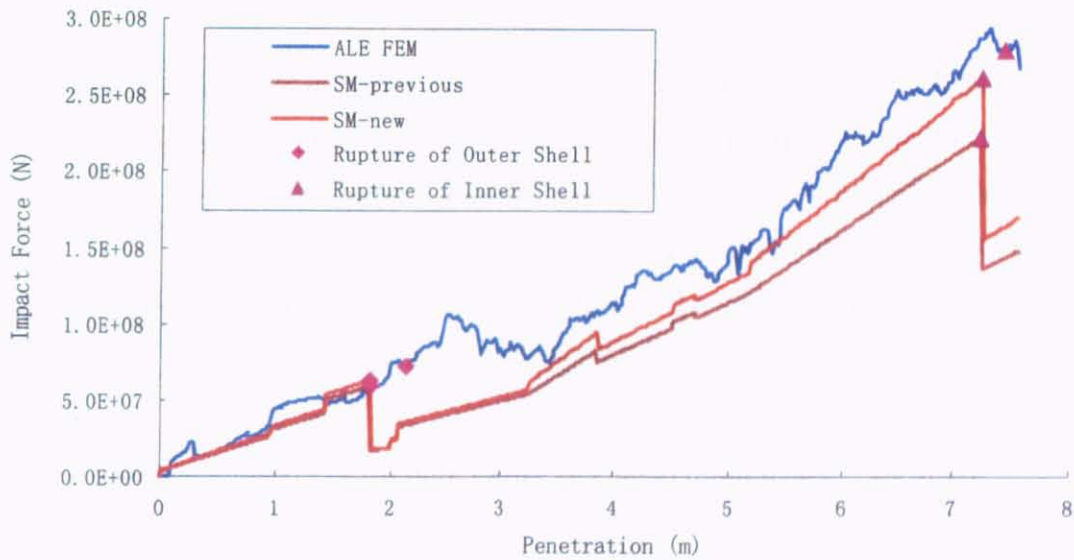


Fig. 8.9 Impact force-penetration curve for different methods

In this collision case, the induced motions of struck ship include sway and roll. Applying the new simplified analytical model, the impact force-penetration curves using previous simplified analytical model, new simplified analytical model and ALE FEM are plotted in Fig. 8.9. At the time of inner shell rupture, the new simplified analytical model can increase 16% impact force compared with the results of previous simplified analytical model.

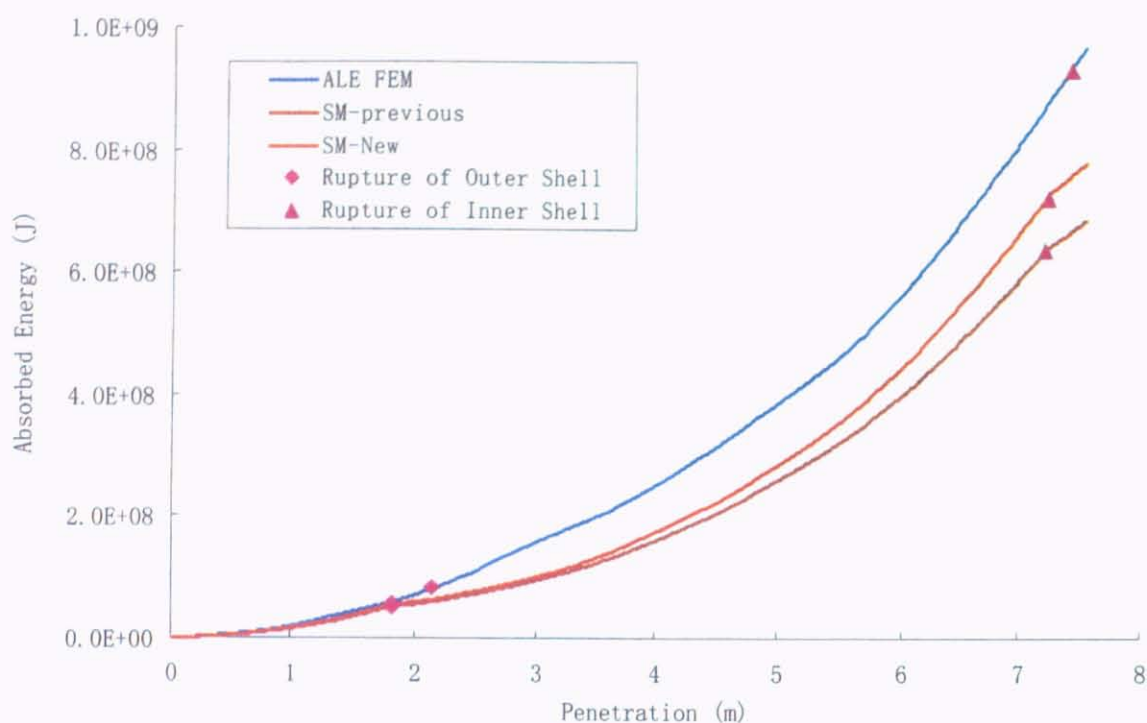


Fig. 8.10 Total absorbed energy-penetration curve for different methods

Fig. 8.10 shows total absorbed energy of struck ship for different methods. In Table 8.2, the penetration depths, impact forces, total absorbed energy at the point of inner shell rupture for previous simplified analytical model, new simplified analytical model and ALE FEM are given. Compared with ALE FEM result, the accuracy of new simplified analytical method is higher than the previous simplified mechanics method.

Table 8.2 Calculation results for struck ship using different methods

	Previous model	Present model	ALE FEM
Time of Inner Shell Rupture(s)	0.924	0.926	0.972
Penetration Depth (m)	7.24	7.26	7.45
Impact Force (MN)	223	262	280
Total absorbed Energy of Struck Ship(MJ)	634	718	926

## 8.5 Benchmark Studies on New Simplified Analytical Model and Nonlinear FEM for Ship Side Collisions

In new simplified analytical model, three control variables, which are striking velocity ( $V_a$ ), collision angle ( $\theta$ ) and collision location ( $\frac{x}{L}$ ), can be taken into account. Compared with ALE FE simulation results for a variety of ship side collision scenarios in chapter 7, benchmark studies on present model are performed in this section.

- **Effect of striking velocity**

Using new simplified analytical model, numerical calculations for side collision between 72,000 tonne oil tanker and 293,000 tonne VLCC are performed for the initial striking velocity with 16knot, 14knot and 12knot. The principal dimensions and material properties of the striking/struck ships are the same as those of the model in chapter 5, which are shown in Table 5.1, 5.2.

The impact force-penetration curves for different striking velocities are plotted in Fig. 8.11. Fig. 8.12 shows total absorbed energy-penetration of struck ship for different striking velocities.

In Table 8.3, the penetration depths, impact forces, total absorbed energy at the point of inner shell rupture for different striking velocities are given.

Table 8.3 Calculation results for struck ship for different striking velocities

Striking Velocity (knot)	16		14		12	
	Present model	ALE FEM	Present model	ALE FEM	Present model	ALE FEM
Time of Inner Shell Rupture(s)	0.62	0.58	0.78	0.97	1.09	1.20
Penetration (m)	5.99	5.65	5.97	5.95	6.00	5.84
Impact Force (MN)	215	220	217	232	223	218
Total absorbed Energy (MJ)	486	623	481	643	491	625

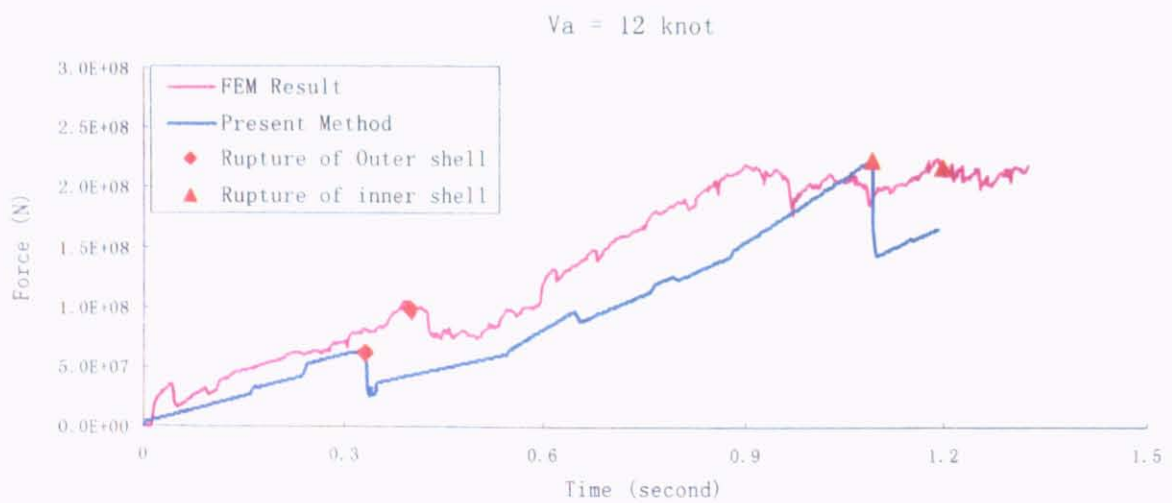
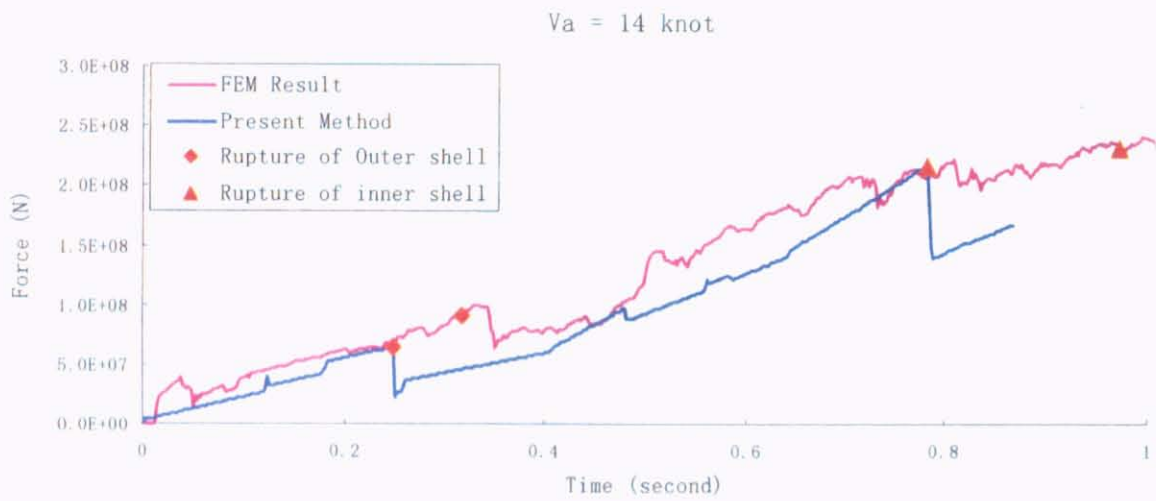
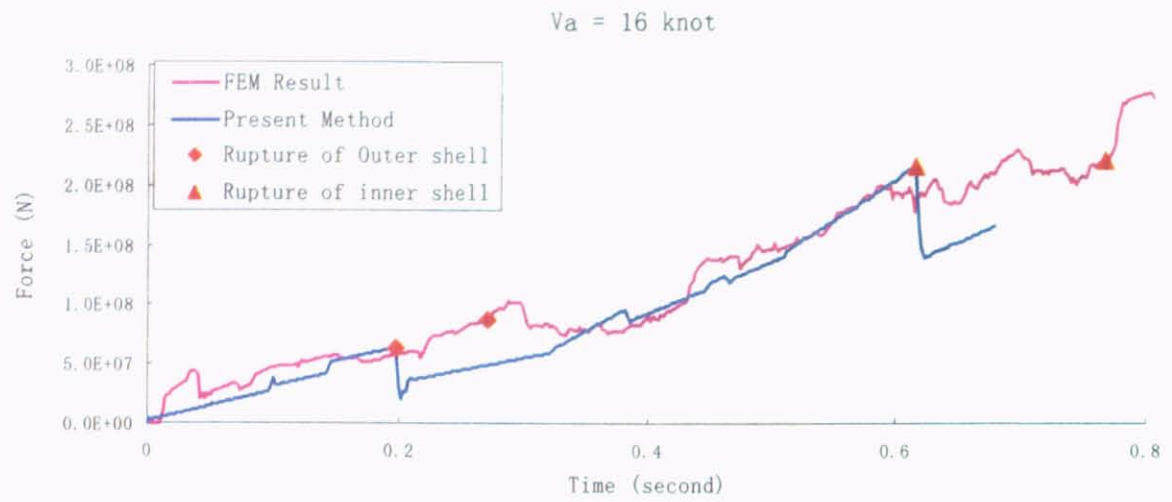


Fig. 8.11 Impact force-penetration curve for different striking velocities

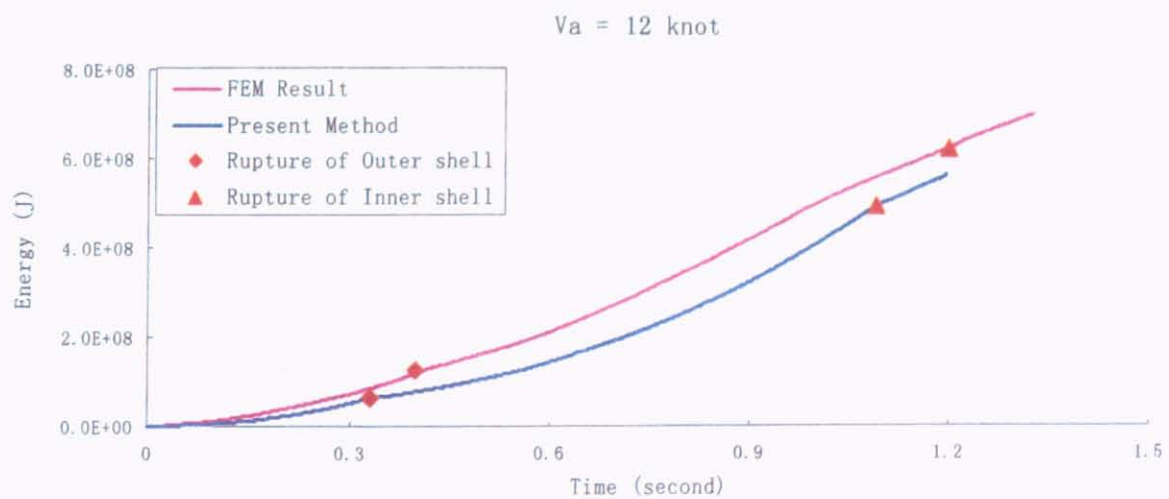
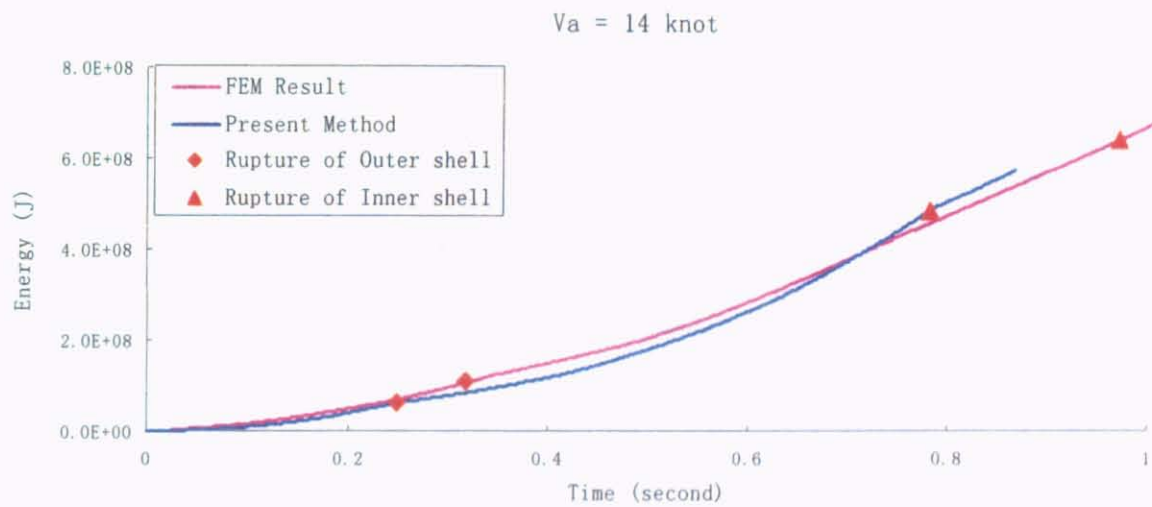
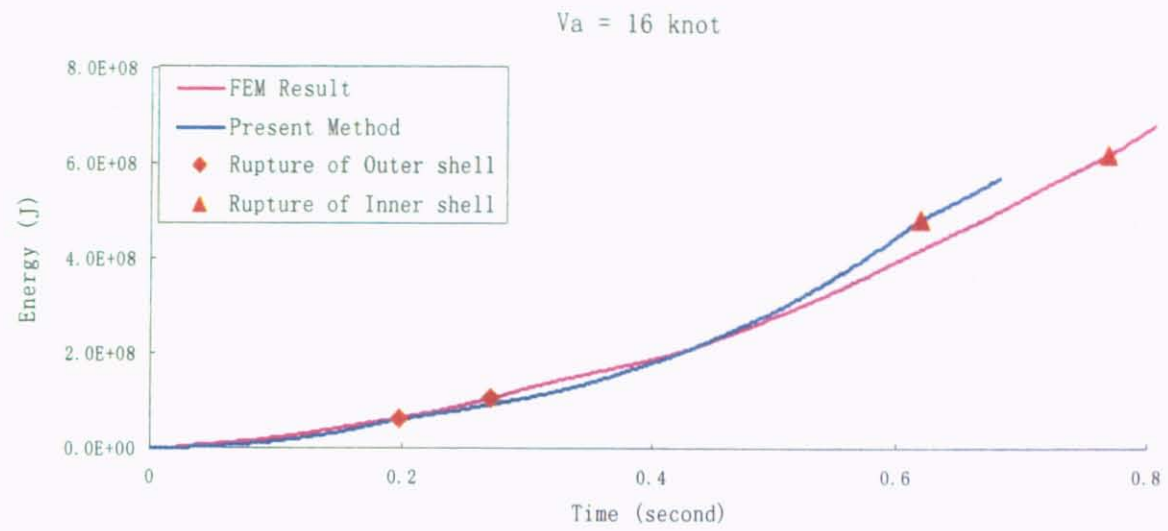


Fig. 8.12 Absorbed energy-penetration curve for different striking velocities

Numerical calculation results of new simplified analytical model show that the impact force and absorbed internal energy is similar at the time of inner shell rupture for various initial striking velocities, which is the same conclusion derived by ALE FEM simulation results. Compared with FEM result, the accuracy of predicted impact force is better than that of absorbed energy.

- **Effect of collision angle**

In new simplified analytical model, we assume that the component of the impact force ( $F_c(\theta)$ ) along the direction of penetration ( $\delta$ ) is same when the collision angle is  $90^\circ$  (as Fig.8.13 shown):

$$F_c(\theta) = \frac{F_c(90^\circ)}{\sin \theta} \quad (8.63)$$

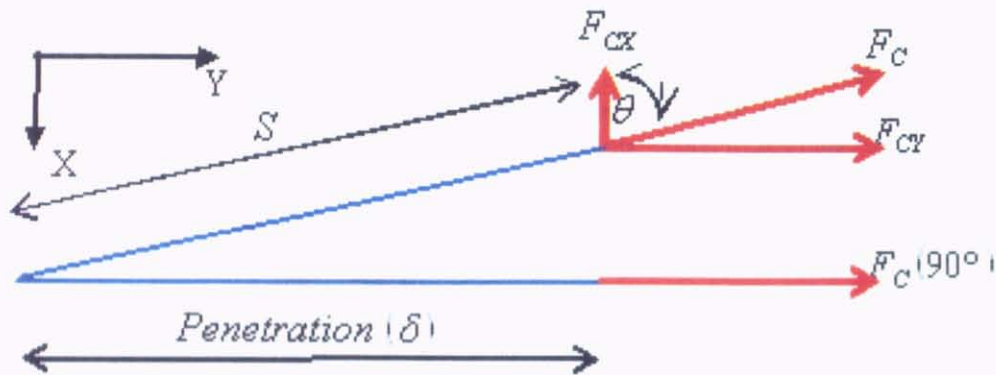


Fig. 8.13 Impact force components for collision angle  $\theta$

Using new simplified analytical model, numerical calculations for side collision between 72,000 tonne oil tanker and 293,000 tonne VLCC are performed for the collision angle with 90 degree, 75 degree and 60 degree. The principal dimensions and material properties of the striking/struck ships are the same as those of the model in chapter 5, which are shown in Table 5.1, 5.2.

The impact force-penetration curves for different collision angles are plotted in Fig. 8.14. Fig. 8.15 shows total absorbed energy-energy of struck ship for different collision angles. In Table 8.4, the penetration depths, impact forces, total absorbed energy at the point of inner shell rupture for different collision angles are given.

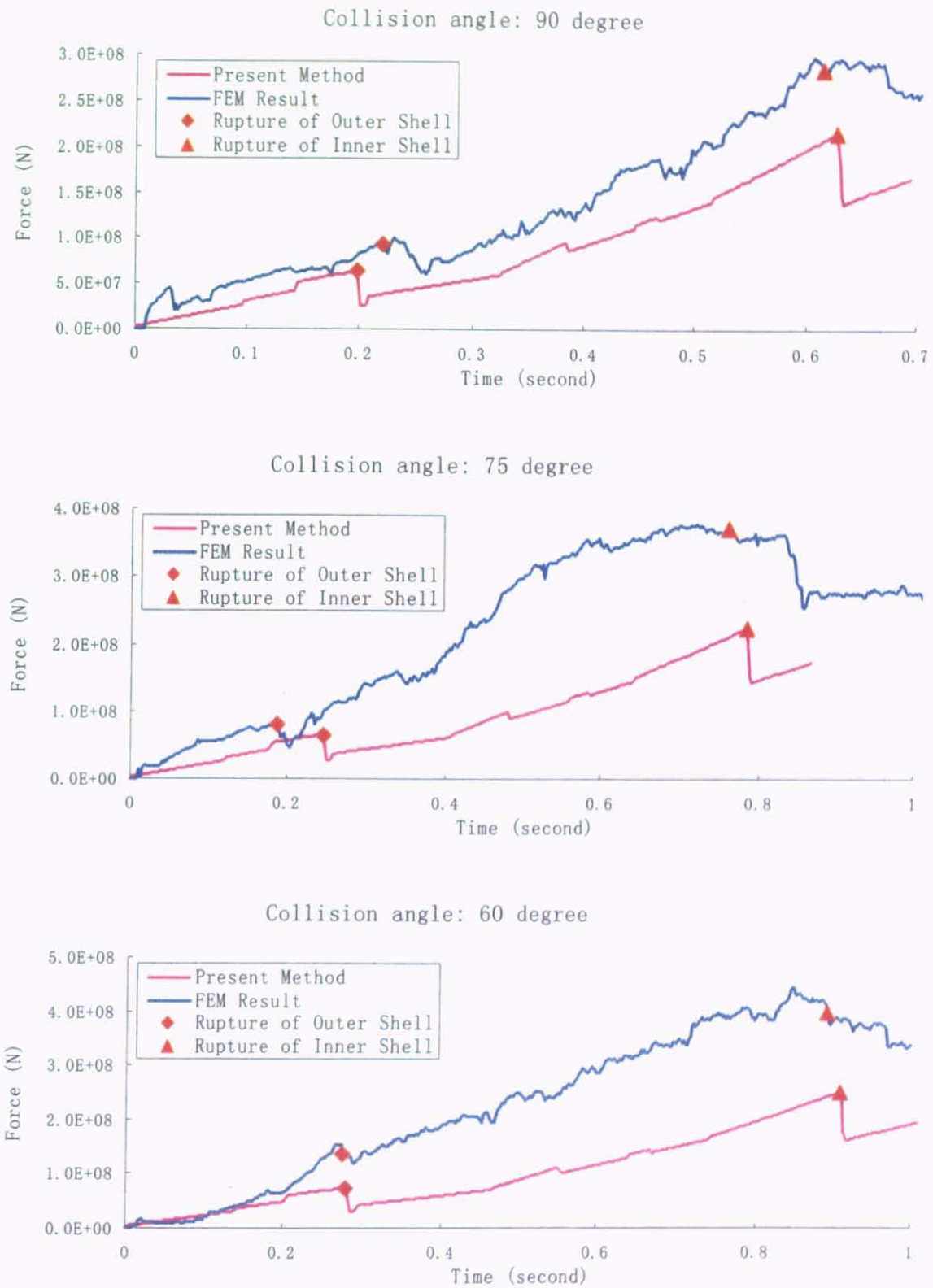


Fig. 8.14 Impact force-penetration curve for different collision angles

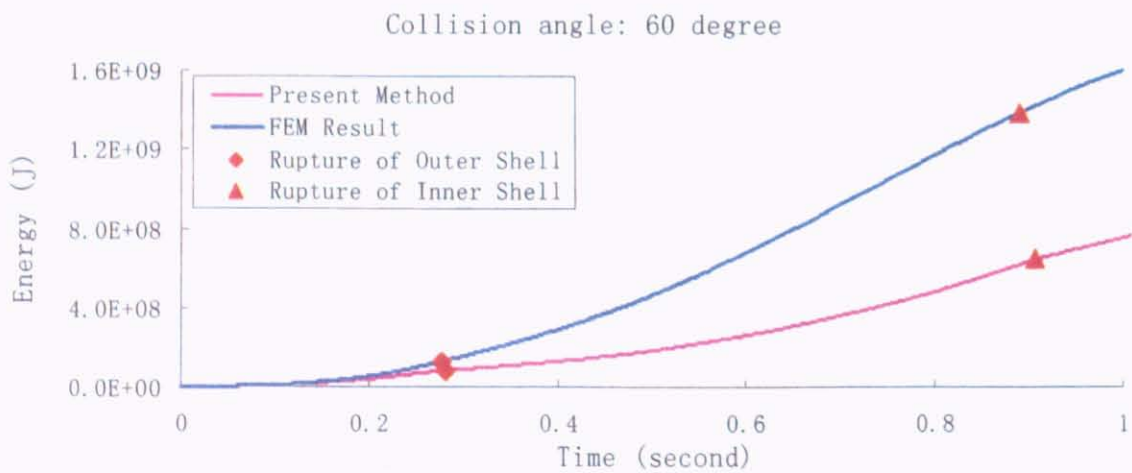
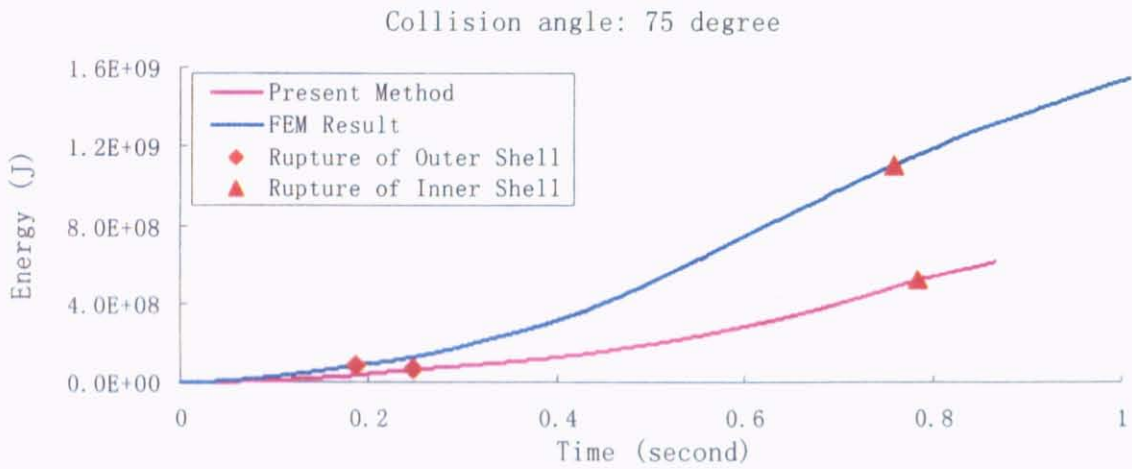
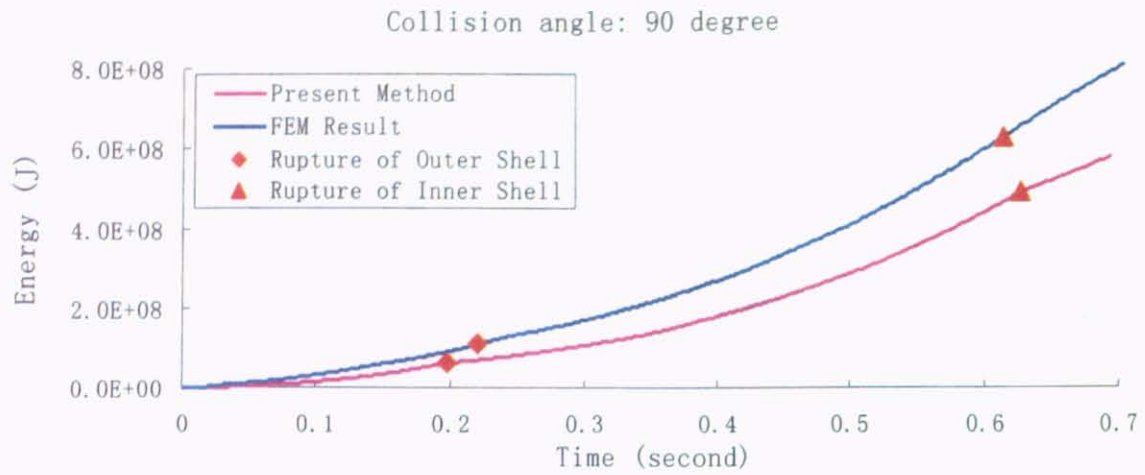


Fig. 8.15 Absorbed energy-penetration curve for different collision angles

Table 8.4 Calculation results for struck ship for different collision angles

Collision angle (degree)	90		75		60	
	Present model	ALE FEM	Present model	ALE FEM	Present model	ALE FEM
Time of Inner Shell Rupture(s)	0.63	0.61	0.78	0.76	0.91	0.89
Penetration (m)	5.99	5.83	6.00	7.54	6.01	7.04
Impact Force (MN)	217	287	224	372	252	399
Total absorbed Energy (MJ)	487	646	522	1107	650	1387

From the comparison in Table 8.4, it is seen that the new simplified analytical method can consider the effect of collision angle on impact force and absorbed energy reasonable. Smaller collision angle leads to larger resistance of side structures at the time of inner shell rupture. To impact force and total absorbed energy at the point of inner shell rupture, the agreement for collision angle 90 degree is excellent. The difference for collision angle 60 is relatively large. Similar to the conclusion derived by ALE FEM simulation results, the perpendicular collision ( $90^\circ$ ) causes the maximum impact force and absorbed energy of side structures at the same collision time.

- **Effect of collision location**

Using new simplified analytical model, numerical calculations for side collision between 350,000 tonne VLCC and 293,000 tonne VLCC are performed for the collision location ( $\frac{x}{L}$ ) with 0.2, 0.5 and 0.65. The principal dimensions and material properties of the striking/struck ships are the same as those of the model in chapter 6.

The impact force-penetration curves for different collision locations are plotted in Fig. 8.16. Fig. 8.17 shows total absorbed energy-energy of struck ship for different collision location. In Table 8.5, the penetration depths, impact forces, total absorbed energy at the point of inner shell rupture for different collision locations are given.

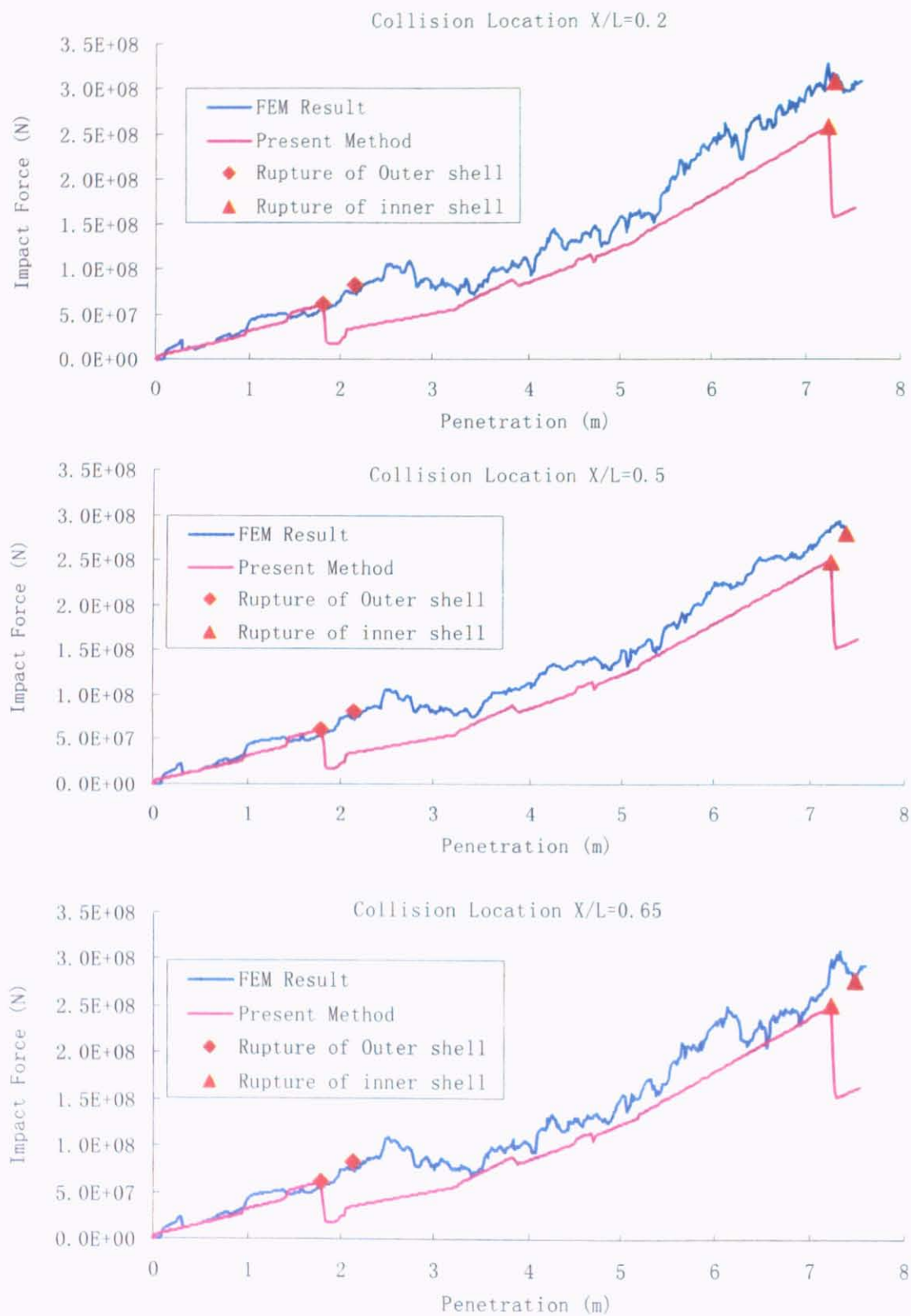


Fig. 8.16 Impact force-penetration curve for different collision locations

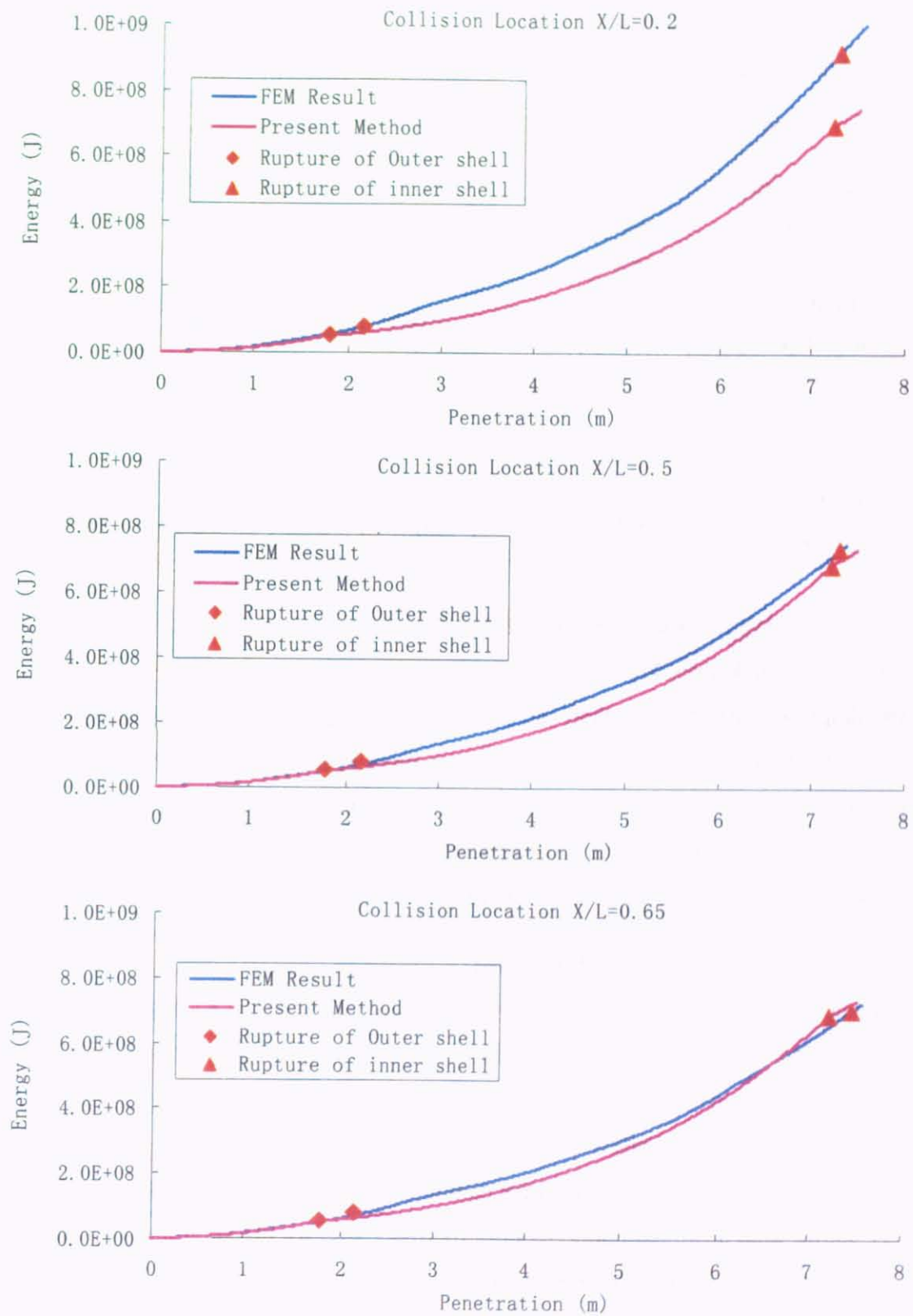


Fig. 8.17 Absorbed energy-penetration curve for different collision locations

Table 8.5 Calculation results for struck ship for different collision locations

Collision location (X/L)	0.2		0.5		0.65	
	Present model	ALE FEM	Present model	ALE FEM	Present model	ALE FEM
Time of Inner Shell Rupture(s)	1.062	0.904	1.062	0.892	1.062	0.882
Penetration (m)	7.231	7.299	7.233	7.327	7.232	7.480
Impact Force (MN)	258.6	308.2	248.5	294.6	251.0	277.34
Total absorbed Energy (MJ)	697.0	921.0	683.5	736.3	686.9	705.3

In new simplified analytical model, different collision location cause different extent yaw motion of struck ship. From the comparison in Fig. 8.17 and Table 8.5, it is seen that the new simplified analytical method can consider the effect of collision location on impact force and absorbed energy reasonable. To impact force at the point of inner shell rupture, the agreement for collision location 0.65 is excellent. The difference for collision location 0.2 is relatively large. The agreement for impact force at rupture of outer and inner shell for different collision location is better than that of total absorbed energy.

- **Effect of different striking ships**

Using new simplified analytical model, numerical calculations for different striking ship are performed for 72,000 tonne oil tanker, 350,000 tonne VLCC oil tanker for laden condition and ballast condition. The principal dimensions and material properties of the striking/struck ships are the same as those of the model in chapter 5 and 6.

The impact force-penetration curves for different striking ships are plotted in Fig. 8.18. Fig. 8.19 shows total absorbed energy-energy of struck ship for different striking ships. In Table 8.6, the penetration depths, impact forces, total absorbed energy at the point of inner shell rupture for different ships are given.

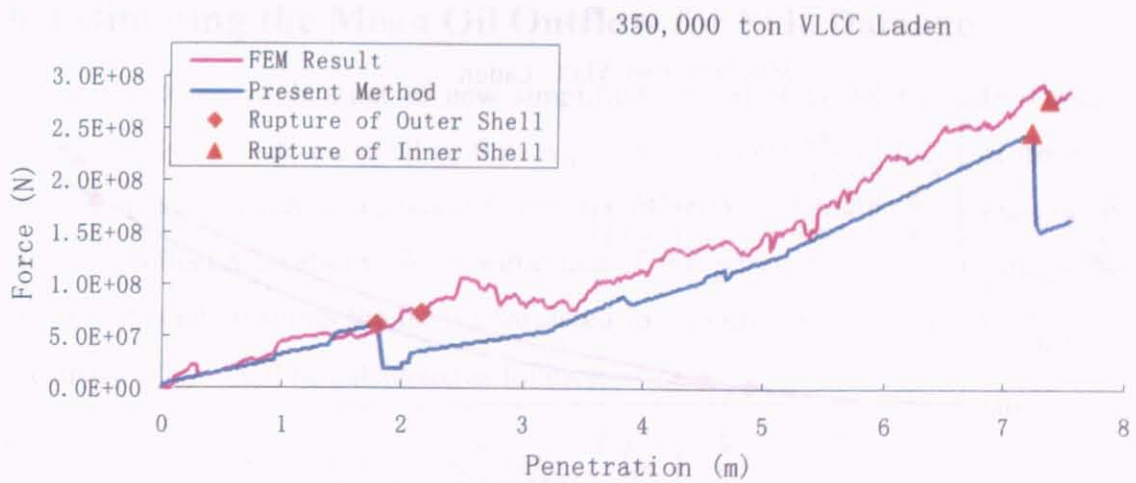
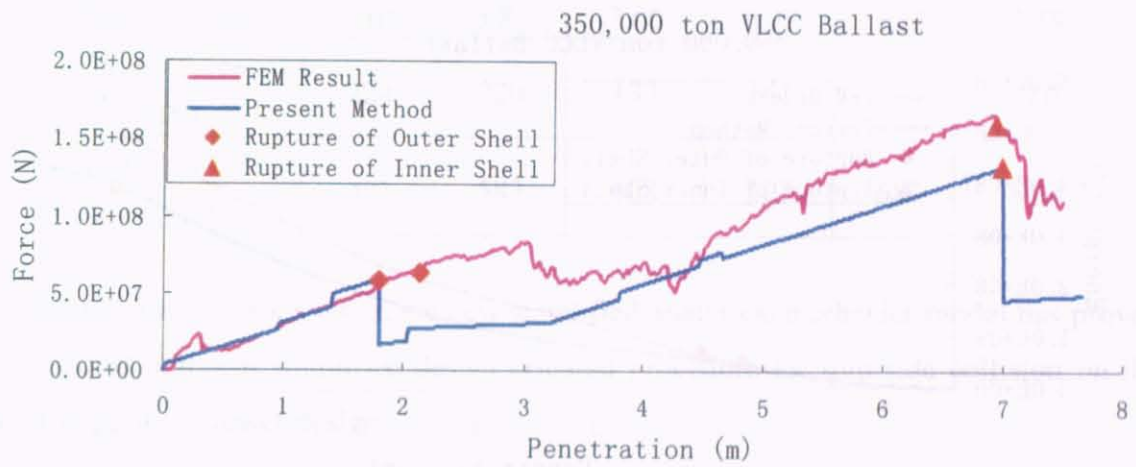
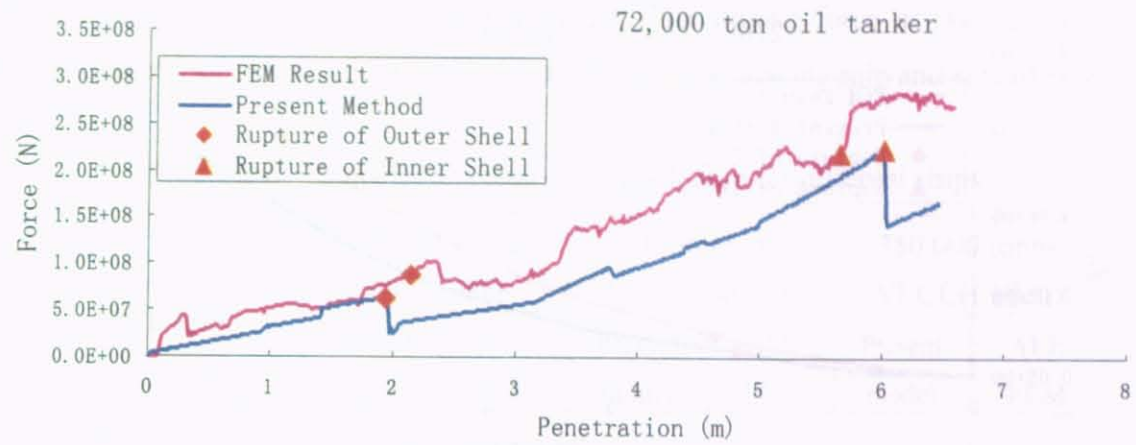


Fig. 8.18 Impact force-penetration curve for different ships

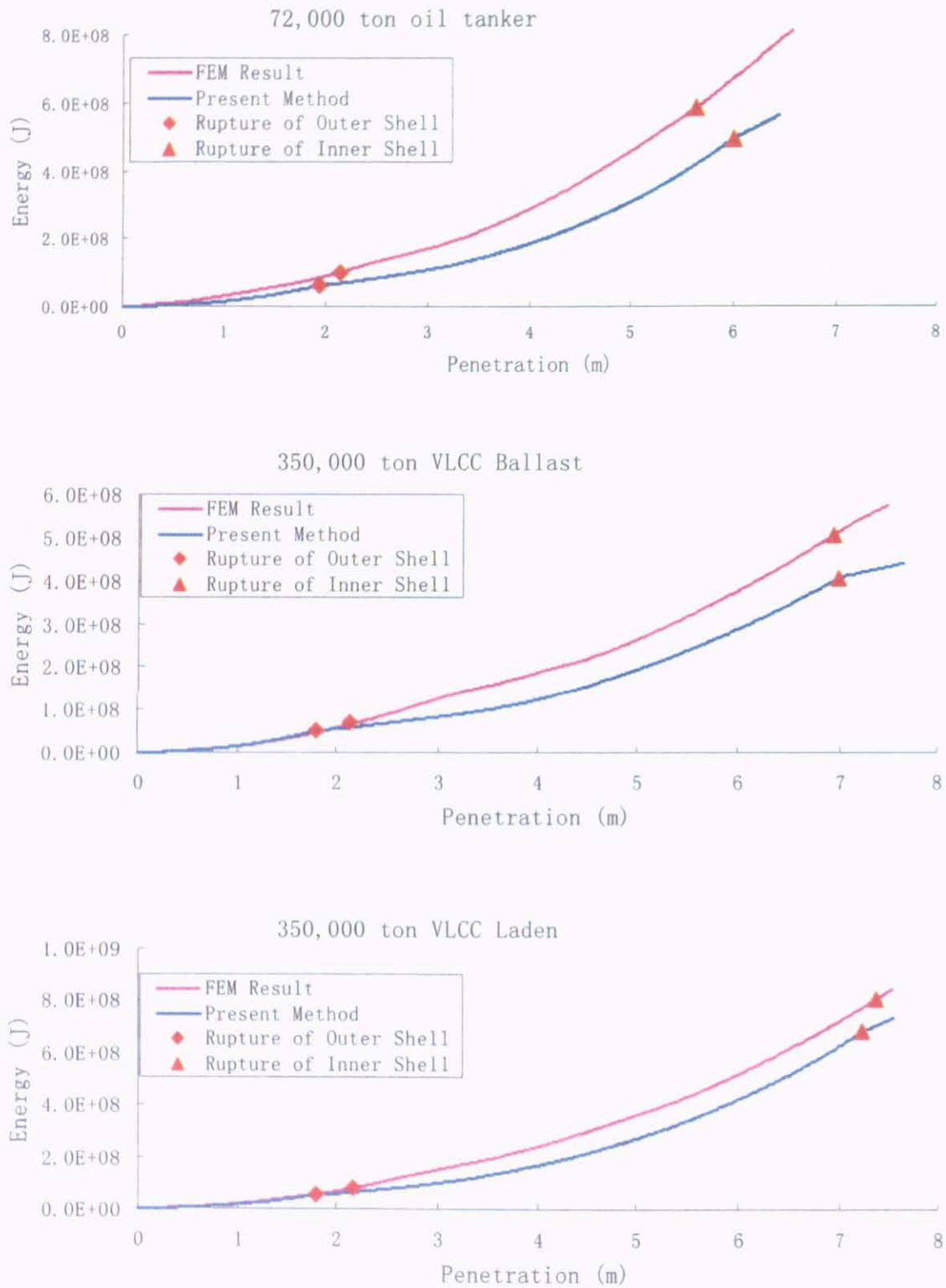


Fig. 8.19 Absorbed energy-penetration curve for different ships

From the comparison in Table 8.6, it is seen that the general agreement between the present method and FEM results is reasonable. The impact force as well as the total absorbed energy of side structures varied with the mass of striking ship and size of bow.

Table 8.6 Calculation results for struck ship for different ships

Ship Type	72,000 tonne Oil Tanker		350,000 tonne VLCC(Ballast)		350,000 tonne VLCC(Laden)	
	Present model	ALE FEM	Present model	ALE FEM	Present model	ALE FEM
Time of Inner Shell Rupture(s)	0.798	0.768	0.910	0.916	1.062	0.972
Penetration (m)	6.01	5.83	7.00	6.95	7.23	7.38
Impact Force (MN)	224	220	133	161	248	280
Total absorbed Energy (MJ)	493	583	408	507	683	810

Through benchmark studies, the new simplified analytical mechanics model has proved to a rational practical simplified design oriented procedure for ship side collision on the initial stage of oil tanker design

## 8.6 Estimating the Mean Oil Outflow for Side Damage

In this section, application of new simplified analytical model for side damage to probabilistically estimate mean oil outflow is presented. Here the probability of penetrating cargo tank in case of side collisions considers the effect of the striking velocities, collision angles and collision locations. We assume that if the penetration between the colliding ships exceeds certain limits the oil is considered to be outflow. The mean oil outflow for side damage  $O_{MS}$  shall be calculated as follows:

$$O_{MS} = \sum P_s \left( V_a, \theta, \frac{x}{L} \right) \cdot O_s \quad (m^3) \quad (8.64)$$

Where

$P_s$  probability of penetrating cargo tank from side damage considering the effect of striking velocity ( $V_o$ ), collision angle ( $\theta$ ) and location ( $\frac{x}{L}$ ).

$O_s$  oil outflow from side damage to cargo tank, which is assumed equal to the total volume in cargo tank, in  $m^3$ .

Based on the ship collision database of Japan, Endo et al (2005) assume that probability density function (PDF) of the striking velocities is uniform distribution. For VLCC of striking ship, the range of striking velocities is from 0 to 16 knot with 1 knot incremental (as Fig.8.17 shown). Similar to the striking velocity case, the PDF of collision angles is assumed the uniform distribution (from 0 degree to 180 degree with 5 degree incremental, as Fig.8.18 shown). The PDF of collision location specifies a uniform constant value over the entire length of the struck ship (as Fig.8.19 shown).

Based on the Sandia National Laboratory Report(1998) and USCG ship casualty data(1991), Brown(2002) uses a Normal distribution ( $\mu = 90$  degrees,  $\sigma = 28.97$  degrees) to describe the PDF of collision angle. The PDF of striking velocity is plotted in Fig.8.18.

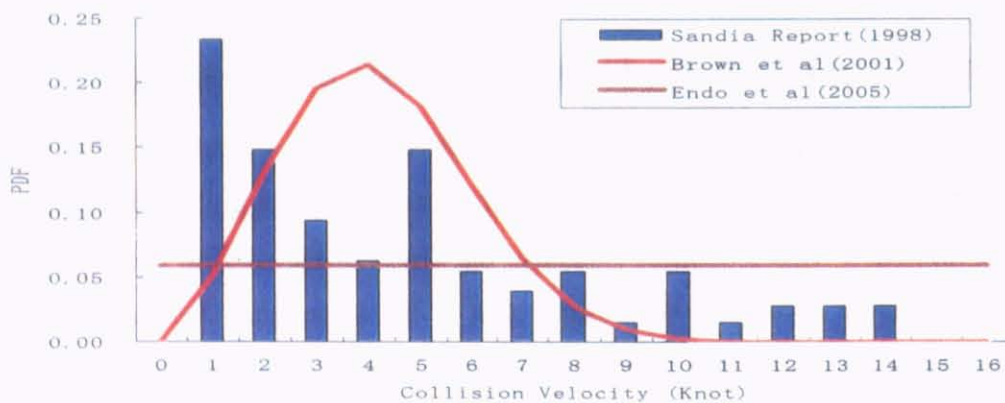


Fig. 8.20 Striking ship velocity PDF

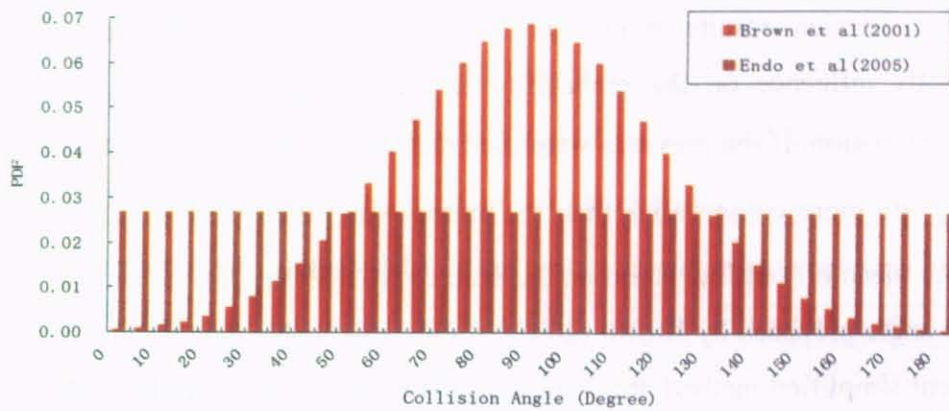


Fig. 8.21 Collision angle PDF

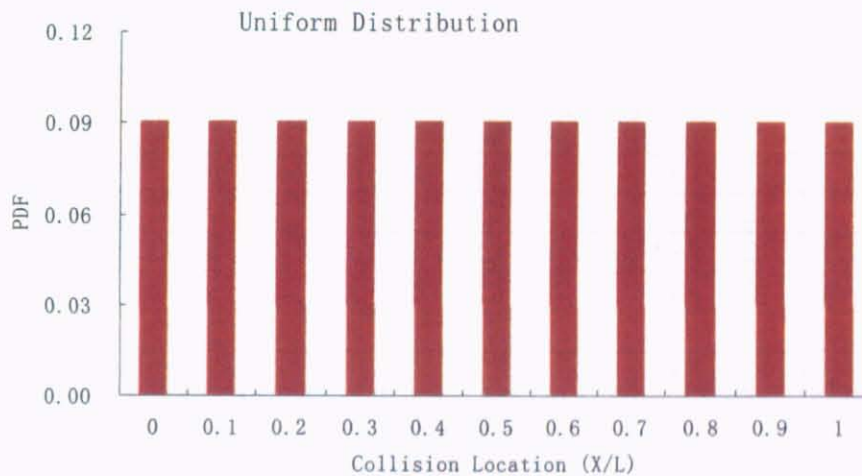


Fig. 8.22 Collision location PDF

Application of previous simplified method and present simplified method, a series of calculations are performed for 293,000 tonne VLCC struck by 350,000 tonne VLCC. Table 8.6 and 8.7 provide the map of penetrating cargo tank for different collision angles and striking velocities, respectively.

The probability of penetrating cargo tank from side damage ( $P_s$ ) and the mean oil outflow ( $O_{MS}$ ) are estimated by using previous simplified method, present method. Based on the revised draft of MARPOL 73/78 Annex I (IMO BLG 2003), a probabilistic method estimating oil outflow from a struck tanker is also performed. The results are presented in table 8.8 comparing to each other.

The estimation results show that PDF of striking velocity and collision angle significantly influence on the probability of penetrating cargo tank and the mean oil outflow estimation. If the previous simplified method is used, the estimating results of  $P_s$  and  $O_{MS}$  are more conservative than the present simplified method. Compared with the estimation mean oil outflow based on the draft of MARPOL, PDF of striking velocity and collision angle proposed by Brown (2002) is more reasonable than Endo(2005). Combined the present simplified method and PDF of collision angles and striking velocity proposed by Brown (2002), a good estimating result of  $P_s$  and  $O_{MS}$  can be obtained.

Table 8.7 Map of penetrating cargo tank using present method

$V_a = 16 \text{ knot}$		Collision Angle (degree)												
		25	30	35	40	45	50-60	65-110	115-125	130	135-140	145	150	155
Collision Location(X/L)	0						○	○						
	0.1			○	○	○	○	○	○	○				
	0.2			○	○	○	○	○	○	○	○			
	0.3		○	○	○	○	○	○	○	○	○	○	○	
	0.4		○	○	○	○	○	○	○	○	○	○	○	○
	0.5	○	○	○	○	○	○	○	○	○	○	○	○	○
	0.6		○	○	○	○	○	○	○	○	○	○	○	○
	0.7			○	○	○	○	○	○	○	○	○	○	
	0.8					○	○	○	○	○	○	○	○	
	0.9						○	○	○	○	○	○		
	1							○	○					

$V_a = 15 \text{ knot}$		Collision Angle (degree)													
		25	30	35	40	45	50-55	60	65-110	115-125	130	135-140	145	150	155
Collision Location(X/L)	0							○	○						
	0.1				○	○	○	○	○	○	○				
	0.2			○	○	○	○	○	○	○	○	○			
	0.3			○	○	○	○	○	○	○	○	○	○		
	0.4		○	○	○	○	○	○	○	○	○	○	○	○	
	0.5	○	○	○	○	○	○	○	○	○	○	○	○	○	○
	0.6		○	○	○	○	○	○	○	○	○	○	○	○	
	0.7			○	○	○	○	○	○	○	○	○	○		
	0.8					○	○	○	○	○	○	○	○		
	0.9						○	○	○	○	○	○			
	1								○	○					

$V_a = 14 \text{ knot}$		Collision Angle (degree)										
		30	35-40	45	50-55	60	65-110	115-125	130	135-140	145	150
Collision Location(X/L)	0					○	○					
	0.1		○	○	○	○	○	○	○			
	0.2		○	○	○	○	○	○	○	○		
	0.3		○	○	○	○	○	○	○	○	○	
	0.4	○	○	○	○	○	○	○	○	○	○	○
	0.5	○	○	○	○	○	○	○	○	○	○	○
	0.6	○	○	○	○	○	○	○	○	○	○	○
	0.7		○	○	○	○	○	○	○	○	○	○
	0.8			○	○	○	○	○	○	○	○	
	0.9				○	○	○	○	○	○	○	
	1						○	○				

“○” means the penetration of cargo tank occurred

$V_a = 13 \text{ knot}$		Collision Angle (degree)										
		30	35-40	45-55	60	65-75	80-100	105-115	120	125-135	140-145	150
Collision Location(X/L)	0					○	○					
	0.1			○	○	○	○	○	○			
	0.2		○	○	○	○	○	○	○	○		
	0.3	○	○	○	○	○	○	○	○	○	○	
	0.4	○	○	○	○	○	○	○	○	○	○	○
	0.5	○	○	○	○	○	○	○	○	○	○	○
	0.6	○	○	○	○	○	○	○	○	○	○	○
	0.7		○	○	○	○	○	○	○	○	○	○
	0.8			○	○	○	○	○	○	○	○	
	0.9				○	○	○	○	○	○		
	1						○	○				

$V_a = 12 \text{ knot}$		Collision Angle (degree)										
		35	40	45	50-55	60-75	80-105	110-115	120	125-135	140	145
Collision Location(X/L)	0					○	○					
	0.1				○	○	○	○	○			
	0.2		○	○	○	○	○	○	○	○		
	0.3	○	○	○	○	○	○	○	○	○	○	
	0.4	○	○	○	○	○	○	○	○	○	○	○
	0.5	○	○	○	○	○	○	○	○	○	○	○
	0.6	○	○	○	○	○	○	○	○	○	○	○
	0.7		○	○	○	○	○	○	○	○	○	○
	0.8			○	○	○	○	○	○	○	○	
	0.9					○	○	○	○	○		
	1						○	○				

$V_a = 11 \text{ knot}$		Collision Angle (degree)										
		35	40	45-50	55	60-70	75-80	85-95	100-105	110-120	125-135	140
Collision Location(X/L)	0						○	○				
	0.1				○	○	○	○	○	○		
	0.2		○	○	○	○	○	○	○	○	○	
	0.3	○	○	○	○	○	○	○	○	○	○	
	0.4	○	○	○	○	○	○	○	○	○	○	○
	0.5	○	○	○	○	○	○	○	○	○	○	○
	0.6	○	○	○	○	○	○	○	○	○	○	○
	0.7		○	○	○	○	○	○	○	○	○	○
	0.8			○	○	○	○	○	○	○	○	
	0.9					○	○	○	○	○	○	
	1							○	○			

“○” means the penetration of cargo tank occurred

$V_a = 10 \text{ knot}$		Collision Angle (degree)											
		40	45-50	55-60	65-70	75-80	85	90-105	110-115	120-125	130	135	140
Collision Location(X/L)	0							○					
	0.1				○	○	○	○					
	0.2		○	○	○	○	○	○	○	○			
	0.3	○	○	○	○	○	○	○	○	○	○	○	
	0.4	○	○	○	○	○	○	○	○	○	○	○	○
	0.5	○	○	○	○	○	○	○	○	○	○	○	○
	0.6	○	○	○	○	○	○	○	○	○	○	○	○
	0.7		○	○	○	○	○	○	○	○	○	○	○
	0.8			○	○	○	○	○	○	○	○		
	0.9					○	○	○	○				
	1						○	○					

$V_a = 9 \text{ knot}$		Collision Angle (degree)									
		45	50	55	60-70	75-85	90	95-105	110-120	125-130	135
Collision Location(X/L)	0										
	0.1					○	○				
	0.2			○	○	○	○	○	○		
	0.3	○	○	○	○	○	○	○	○	○	
	0.4	○	○	○	○	○	○	○	○	○	○
	0.5	○	○	○	○	○	○	○	○	○	○
	0.6	○	○	○	○	○	○	○	○	○	○
	0.7		○	○	○	○	○	○	○	○	○
	0.8				○	○	○	○	○	○	
	0.9						○	○			
	1										

$V_a = 8 \text{ knot}$		Collision Angle (degree)												
		55	60	65-70	75	80	85	90	95	100	105	110-115	120	125
Collision Location(X/L)	0													
	0.1					○	○	○						
	0.2				○	○	○	○	○					
	0.3			○	○	○	○	○	○	○	○	○		
	0.4		○	○	○	○	○	○	○	○	○	○	○	
	0.5	○	○	○	○	○	○	○	○	○	○	○	○	○
	0.6		○	○	○	○	○	○	○	○	○	○	○	
	0.7		○	○	○	○	○	○	○	○	○	○		
	0.8						○	○	○	○	○			
	0.9							○	○	○				
	1													

“○” means the penetration of cargo tank occurred

$V_a = 7 \text{ knot}$		Collision Angle (degree)								
		70	75	80	85	90	95	100	105	110
Collision Location(X/L)	0									
	0.1			○	○	○				
	0.2		○	○	○	○	○			
	0.3	○	○	○	○	○	○	○		
	0.4	○	○	○	○	○	○	○	○	
	0.5	○	○	○	○	○	○	○	○	○
	0.6		○	○	○	○	○	○	○	○
	0.7			○	○	○	○	○	○	
	0.8				○	○	○	○	○	
	0.9				○	○	○			
	1									

$V_a = 6 \text{ knot}$		Collision Angle (degree)				
		80	85	90	95	100
Collision Location(X/L)	0.2					
	0.3		○			
	0.4		○	○		
	0.5		○	○	○	
	0.6			○	○	
	0.7				○	
	0.8					
	0.9					
	1					

Table 8.8 Map of penetrating cargo tank using previous method

$V_a = 16 \text{ knot}$		Collision Angle (degree)													
		25	30	35	40	45	50-60	65-110	115-125	130	135-140	145	150	155	160
Collision Location(X/L)	0						○	○							
	0.1			○	○	○	○	○	○	○					
	0.2			○	○	○	○	○	○	○	○				
	0.3		○	○	○	○	○	○	○	○	○	○	○		
	0.4		○	○	○	○	○	○	○	○	○	○	○	○	
	0.5	○	○	○	○	○	○	○	○	○	○	○	○	○	○
	0.6		○	○	○	○	○	○	○	○	○	○	○	○	
	0.7			○	○	○	○	○	○	○	○	○	○	○	
	0.8					○	○	○	○	○	○	○	○		
	0.9						○	○	○	○	○	○			
	1							○	○						

“○” means the penetration of cargo tank occurred

$V_a = 15 \text{ knot}$		Collision Angle (degree)													
		25	30	35-40	45	50-55	60	65-110	115-125	130	135-140	145	150	155	160
Collision Location(X/L)	0						○	○							
	0.1			○	○	○	○	○	○	○					
	0.2			○	○	○	○	○	○	○	○				
	0.3		○	○	○	○	○	○	○	○	○	○	○		
	0.4		○	○	○	○	○	○	○	○	○	○	○	○	
	0.5	○	○	○	○	○	○	○	○	○	○	○	○	○	○
	0.6		○	○	○	○	○	○	○	○	○	○	○	○	
	0.7			○	○	○	○	○	○	○	○	○	○	○	
	0.8				○	○	○	○	○	○	○	○	○		
	0.9					○	○	○	○	○	○	○			
	1							○	○						

$V_a = 14 \text{ knot}$		Collision Angle (degree)													
		25	30	35-40	45	50-55	60	65-110	115-125	130	135-140	145	150	155	160
Collision Location(X/L)	0						○	○							
	0.1			○	○	○	○	○	○	○					
	0.2			○	○	○	○	○	○	○	○				
	0.3		○	○	○	○	○	○	○	○	○	○	○		
	0.4		○	○	○	○	○	○	○	○	○	○	○		
	0.5	○	○	○	○	○	○	○	○	○	○	○	○	○	○
	0.6		○	○	○	○	○	○	○	○	○	○	○	○	
	0.7			○	○	○	○	○	○	○	○	○	○	○	
	0.8				○	○	○	○	○	○	○	○	○		
	0.9					○	○	○	○	○	○	○			
	1							○	○						

$V_a = 13 \text{ knot}$		Collision Angle (degree)										
		30	35-40	45-55	60	65-75	80-100	105-115	120	125-135	140-145	150
Collision Location(X/L)	0					○	○					
	0.1			○	○	○	○	○	○			
	0.2		○	○	○	○	○	○	○	○		
	0.3	○	○	○	○	○	○	○	○	○	○	
	0.4	○	○	○	○	○	○	○	○	○	○	○
	0.5	○	○	○	○	○	○	○	○	○	○	○
	0.6	○	○	○	○	○	○	○	○	○	○	○
	0.7		○	○	○	○	○	○	○	○	○	○
	0.8			○	○	○	○	○	○	○	○	
	0.9				○	○	○	○	○	○		
	1						○	○				

“○” means the penetration of cargo tank occurred

$V_a = 12 \text{ knot}$		Collision Angle (degree)													
		30	35	40-45	50	55	60-70	75-105	110	115-120	125-130	135	140	145	150
Collision Location(X/L)	0						○	○							
	0.1				○	○	○	○	○	○					
	0.2		○	○	○	○	○	○	○	○	○	○			
	0.3	○	○	○	○	○	○	○	○	○	○	○	○		
	0.4	○	○	○	○	○	○	○	○	○	○	○	○	○	○
	0.5	○	○	○	○	○	○	○	○	○	○	○	○	○	○
	0.6	○	○	○	○	○	○	○	○	○	○	○	○	○	○
	0.7		○	○	○	○	○	○	○	○	○	○	○	○	
	0.8			○	○	○	○	○	○	○	○	○	○		
	0.9					○	○	○	○	○	○				
	1							○	○						

$V_a = 11 \text{ knot}$		Collision Angle (degree)												
		30-35	40	45-50	55	60-70	75-80	85-95	100-105	110-120	125-135	140	145	150
Collision Location(X/L)	0						○	○						
	0.1				○	○	○	○	○	○				
	0.2		○	○	○	○	○	○	○	○	○			
	0.3	○	○	○	○	○	○	○	○	○	○	○		
	0.4	○	○	○	○	○	○	○	○	○	○	○	○	○
	0.5	○	○	○	○	○	○	○	○	○	○	○	○	○
	0.6	○	○	○	○	○	○	○	○	○	○	○	○	○
	0.7		○	○	○	○	○	○	○	○	○	○	○	○
	0.8			○	○	○	○	○	○	○	○	○	○	
	0.9					○	○	○	○	○	○			
	1							○	○					

$V_a = 10 \text{ knot}$		Collision Angle (degree)													
		35	40	45-50	55-60	65-70	75-80	85	90-105	110-115	120-125	130	135	140	145
Collision Location(X/L)	0								○						
	0.1					○	○	○	○						
	0.2			○	○	○	○	○	○	○	○				
	0.3		○	○	○	○	○	○	○	○	○	○	○		
	0.4	○	○	○	○	○	○	○	○	○	○	○	○	○	
	0.5	○	○	○	○	○	○	○	○	○	○	○	○	○	○
	0.6		○	○	○	○	○	○	○	○	○	○	○	○	○
	0.7			○	○	○	○	○	○	○	○	○	○	○	
	0.8				○	○	○	○	○	○	○	○			
	0.9						○	○	○	○					
	1							○	○						

“○” means the penetration of cargo tank occurred

$V_a = 9 \text{ knot}$		Collision Angle (degree)											
		40	45	50	55	60-70	75-85	90	95-105	110-120	125-130	135	140
Collision Location(X/L)	0												
	0.1						○	○					
	0.2				○	○	○	○	○	○			
	0.3		○	○	○	○	○	○	○	○	○		
	0.4	○	○	○	○	○	○	○	○	○	○	○	
	0.5	○	○	○	○	○	○	○	○	○	○	○	○
	0.6		○	○	○	○	○	○	○	○	○	○	○
	0.7			○	○	○	○	○	○	○	○	○	
	0.8					○	○	○	○	○	○		
	0.9							○	○				
	1												

$V_a = 8 \text{ knot}$		Collision Angle (degree)													
		45	50-55	60-70	75	80	85	90	95	100	105	110-120	125	130	135
Collision Location(X/L)	0														
	0.1					○	○	○							
	0.2				○	○	○	○	○						
	0.3		○	○	○	○	○	○	○	○	○	○			
	0.4	○	○	○	○	○	○	○	○	○	○	○	○	○	
	0.5	○	○	○	○	○	○	○	○	○	○	○	○	○	○
	0.6		○	○	○	○	○	○	○	○	○	○	○	○	○
	0.7			○	○	○	○	○	○	○	○	○	○		
	0.8						○	○	○	○	○				
	0.9							○	○	○					
	1														

$V_a = 7 \text{ knot}$		Collision Angle (degree)													
		55	60	65	70	75	80	85-90	95	100	105	110	115-120	125	
Collision Location(X/L)	0														
	0.1						○	○							
	0.2					○	○	○	○	○					
	0.3			○	○	○	○	○	○	○	○				
	0.4	○	○	○	○	○	○	○	○	○	○	○	○		
	0.5	○	○	○	○	○	○	○	○	○	○	○	○	○	○
	0.6		○	○	○	○	○	○	○	○	○	○	○	○	○
	0.7				○	○	○	○	○	○	○	○			
	0.8						○	○	○	○	○				
	0.9							○	○	○					
	1														

“○” means the penetration of cargo tank occurred

$V_a = 6 \text{ knot}$		Collision Angle (degree)								
		70	75	80	85	90	95	100	105	110
Collision Location(X/L)	0									
	0.1					○				
	0.2				○	○	○			
	0.3		○	○	○	○	○	○		
	0.4	○	○	○	○	○	○	○	○	
	0.5	○	○	○	○	○	○	○	○	○
	0.6		○	○	○	○	○	○	○	○
	0.7			○	○	○	○	○	○	
	0.8					○	○	○		
	0.9						○			
	1									

“○” means the penetration of cargo tank occurred

Table 8.9 Estimated mean oil outflow

	Endo's Assumption		Brown's Assumption		IMO BLG (2003)
	Previous Method	Present Method	Previous Method	Present Method	
$P_S$	0.2698	0.2509	0.1901	0.1707	0.1667
$O_{MS}(\text{m}^3)$	4889.2	4546.7	3444.9	3093.4	3020.5

## 8.7 Remarks

The structural designer would be more interested in obtaining a quick tool for assessing a structural design for various accident scenarios. A new simplified analytical model for predicting the impact force and the dissipated energy caused by the side structure damage, transverse vibration and global motions of struck ship is derived. The emphasis has been on the dynamic effect of struck ship during collision procedure.

The new simplified analytical method takes collision velocity, collision angle and collision location into account, and can be used for a wide range of different accident scenarios. Calculations using this method compared satisfactorily with FEM simulation results. Through benchmark studies, the new simplified analytical mechanics model has proved to be a rational practical simplified design oriented procedure for ship side collision on the initial stage of oil tanker design. This method can be easily incorporated into a probability-based framework to properly assess structural performance for a variety of damage scenarios.

Application of new simplified method to mean oil outflow estimation is performed. Combined with PDF assumption of collision angles and striking velocity proposed by Brown, the present method can predict more reasonable result of mean oil outflow.

The new simplified analytical method developed, and the results and insights obtained by the present study should be useful for the rational design of double hull tanker side structures against ship collision to reduce the risk of oil pollution and for the collision resistance evaluation of existing double hull tanker structures.

## **Chapter 9 Conclusion and Suggestions for Future Research**

## 9.1 Conclusion

The main objective of this thesis is to establish the rational procedure to evaluate the structural damage, energy dissipation and striking/struck ship motion correctly in case of liquid products tanker's collision and grounding. The Arbitrary Lagrangian Eulerian (ALE) finite element method, which can consider the fluid-structure interactions of liquid products in tank and surrounding water, can improve the accuracy and reliability of nonlinear finite element method for liquid cargo ship collision and grounding simulation.

The crushing collapse of thin-walled structure elements is the most important failure modes during head-on collision. Since nonlinear finite element method becomes a powerful tool in ship collision and grounding problem analysis, a study on the crushing collapse of stiffened square tube using nonlinear FEM was performed. By comparing the experimental data with FEM simulation results, the new equivalent plate thickness formulas for the longitudinally stiffened square tube, transversely stiffened square tube and orthogonally stiffened square tube are derived, respectively. The predicting result of the mean crushing load using the new equivalent plate thickness formulas is in fair agreement with the experiment data.

To improve computing efficiency, accuracy and stability of nonlinear FEM, the effect of selected parameters on crashworthiness of the single-hull bottom structure is investigated. The effect of boundary condition, different shell element types, the residual stress, the material model, the friction coefficient, the rupture strain was discussed.

Through validation of the ALE FE method on the rectangular tank sloshing experiment and the sway motion of hull experiment, the ALE FE method is proved to be suitable to simulate the fluid-structure interaction of liquid cargo in tank and the surrounding water.

Four different numerical FE models for liquid cargo in tank were performed. The advantage of ALE FE method is that it can accurately simulate the fluid-structure interaction in liquid cargo tank. However the CPU time of this method is much larger than other three FE models. On the other hand, although the Linear Sloshing FE model and Rigid Point Mass model need the shortest CPU time, these FE model underestimate or neglect the hydrodynamic force in liquid cargo tank. Compared to the result of ALE FE model, the Lagrangian FE model is appropriate for predicting the structure behavior of

damaged struck cargo tank accounting for the fluid-structure interaction in liquid cargo tank with reasonable accurateness and a relatively low required CPU time.

Numerical simulation of the ship collision between a 350,000 tonne VLCC collide with a 293,000 tonne double hull VLCC, taking account of both fluid-structure interaction in liquid tank and surrounding water, was performed. In order to improve the accuracy of collision behavior of struck ship under the defined collision scenarios, the effect of collision angle, the striking ship velocity, the mass of the striking ship and struck ship velocity, is discussed.

For a reliable and practical evaluation of structural response of side collision, a new simplified analytical internal mechanics model is proposed, which the dynamic effect including transverse vibration and global motions of struck ship on the impact force are considered. Application of present simplified method, estimation oil outflow in side collision is carried out. The mean oil outflow for side impact can be efficiently and accurately evaluated by assuming appropriate PDF of collision angle, impact location and striking velocity model, and assembling the new simplified analytical model.

## **9.2 Suggestions for Future Research**

Currently, there are no widely or generally accepted ship collision and grounding design standards. Design standards should be developed by the community, which should be consisted of classification societies, international organizations (ISO, IMO), leading researchers in the field, representatives from the ship design and shipbuilding communities, ship owners and operators, professional societies, governments and regulators. We should establish a more rational predictive calculation approaches on crashworthiness and safety. Further in-depth studies are needed to investigate the behavior of aluminum and sandwich panels, and innovative designs that maximize the crashworthiness in an accidental impact.

The primary failure modes of plates in ship collision and grounding, such as tearing of plate, concertina tearing of plate, denting of plate, penetration of plate and stretching of plate, should be studied using nonlinear FEM. The effect of geometry of stiffener or indenter can be investigated deeply.

The reliable automated simulation of the structural failure process up to occurrence of fracture in ship collision and grounding is probably the most challenging task in applying

nonlinear FEM. For typical ship structures the stress state is tri-axial which complicates the development of proper failure criteria for rupture. There are various approaches presented to handle the tri-axial failure criteria for rupture. Still the validation of the available approaches is fairly limited especially with real structural configurations. Especially the proper modeling method of the rupture in nonlinear FE-analysis needs further studies.

The element kill algorithm can be coupled with the stiffness reduction and is sometimes a useful method in simulations of fracture initiation and crack propagation due to the simplicity and cost effectiveness. When an element has reached the failure criterion value, the element is deleted from the calculation. This will often cause elastic stress waves in the structure as the stiffness is suddenly reduced. The so-called mesh-free new numerical methods for simulating crack propagation, where problems with distorted elements are somewhat avoided, should be further studies.

Moreover, the effects of including the welds in simulations of ship collision and grounding should be studied.

# **Appendix**

## A1. The Explicit and Implicit Method

Two types of FE methodologies, implicit and explicit techniques, are used to solve the nonlinear problems.

Define the semi-discrete equation of motion is:

$$M\ddot{x}(t) + F_{\text{int}}(x, \dot{x}) - F_{\text{ext}}(x, t) = 0 \quad (\text{A-1})$$

where

$\ddot{x}, \dot{x}, x$  the acceleration, velocity, coordinate vectors

$M$  the diagonal mass matrix

$F_{\text{ext}}$  the external load and body force vector

$F_{\text{int}}$  the stress divergence vector

In order to briefly explain, assumption that the  $F_{\text{int}}^n$  is linear is made

$$F_{\text{int}}(x, \dot{x}) = C\dot{x} + K\Delta x \quad (\text{A-2})$$

where

$K$  the stiffness matrix

$C$  the damping matrix

$\Delta x$  the displacement vector

Regardless of whether an explicit or implicit integration scheme is used, we require that

$$Q = M\ddot{x} + C\dot{x} + K\Delta x - F_{\text{ext}} = 0 \quad (\text{A-3})$$

Explicit integration trivially satisfies these equations since the calculation of the acceleration guarantees equilibrium

$$\ddot{x}^n = M^{-1}(F_{\text{ext}}^n - F_{\text{int}}^n) = M^{-1}(F_{\text{ext}}^n - C\dot{x} - K\Delta x) \quad (\text{A-4})$$

Using central difference integration, the explicit update of the velocities and coordinates is given by

$$\dot{x}^{n+1/2} = \dot{x}^{n-1/2} + \ddot{x}^n \Delta t^n \quad (\text{A-5})$$

$$x^{n+1} = x^n + \dot{x}^{n+1/2} \Delta t^{n+1/2} \quad (\text{A-6})$$

where

$$\Delta t^{n+1/2} = \frac{(\Delta t^n + \Delta t^{n+1})}{2} \quad (\text{A-7})$$

Stability places a limit on the time size. This step size may be very small and a large number of steps may be required.

For the implicit solution, the residual vector  $Q$  becomes an implicit function of  $x^{n+1}$  only. We seek the vector  $x^{n+1}$  such that

$$Q(x^{n+1}) = 0 \quad (\text{A-8})$$

Assume an approximation  $x_k^{n+1}$  to  $x^{n+1}$  for  $k = 1, 2, 3 \dots$  etc. In the neighborhood of  $x_k^{n+1}$ , the linear approximation to  $Q(x^{n+1}) = 0$  is given by

$$Q(x_k^{n+1}) = Q(x_{k-1}^{n+1}) + J(x_{k-1}^{n+1}) \Delta x_k = 0 \quad (\text{A-9})$$

And iterate for the solution:

$$J(x_{k-1}^{n+1}) = \left. \frac{\partial Q}{\partial x} \right|_{x_{k-1}^{n+1}} \quad (\text{A-10})$$

The Jacobian matrix is expressed as

$$J = M \frac{\partial \ddot{x}}{\partial x} + C \frac{\partial \dot{x}}{\partial x} + K - \frac{\partial F_{ext}}{\partial x} \quad (\text{A-11})$$

The solution of increment in displacement is

$$\Delta x_k = -\left(J(x_{k-1}^{n+1})\right)^{-1} Q(x_{k-1}^{n+1}) \quad (\text{A-12})$$

$$x_k^{n+1} = x_{k-1}^{n+1} + \Delta x_k \quad (\text{A-13})$$

If convergence is not satisfied, the displacement vector is updated

$$x_{k+1}^{n+1} = x_k^{n+1} + \Delta x_{k+1} \quad (\text{A-14})$$

And another iteration is performed.

The implicit solvers are properly applied to static, quasi-static, and dynamic problems with a low frequency content. An advantage of the implicit solver than explicit integration is that the number of load or time steps is typically 100 to 10000 times fewer. The major disadvantage is that the cost per step is unknown since the speed depends mostly on the convergence behavior of the equilibrium iterations which can vary widely from problem to problem. The calculation cost depends largely on the number of equations in system and the computer capacity, especially memory resources.

## A2. The Time Integration

During the solution we loop through the elements to update the stress vector and determine a new time step size by taking the minimum value of all elements

$$\Delta t^{i+1} = \alpha \cdot \min\{\Delta t_1, \Delta t_2, \Delta t_3, \dots, \Delta t_N\} \quad (\text{A-15})$$

where

$\Delta t$  the time step

$N$  the total number of elements

$\alpha$  the scale factor, 0.9 or less

The time step calculation for shell elements is as follows.

$$\Delta t_e = \frac{L_s}{C} \quad (\text{A-16})$$

where

$L_s$  the characteristic length

$$L_s = \frac{A_s}{\max(L_1, L_2, L_3, L_4)} \quad (\text{A-17})$$

$C$  the sound speed

$$C = \sqrt{\frac{E}{\rho(1-\gamma^2)}} \quad (\text{A-18})$$

where

$A_s$  the element area

$L_1, L_2, L_3$  and  $L_4$  the lengths of the element sides

$E$  Young's modulus

$\rho$  the density

$\gamma$  Poisson's ratio

## A3. The Effect of Strain Rate

With the increase in the strain rate, the yield stress of material increases, while the ductility is reduced. The most commonly adopted formula to deal with the strain rate sensitivity on yield stress is one proposed by Cowper and Symonds (1957), which is given by

$$\frac{\sigma_{yd}}{\sigma_y} = 1.0 + \left( \frac{\dot{\epsilon}}{C} \right)^{1/q} \quad (\text{A-19})$$

where

$C, q$  coefficients to be determined based on test data

$\sigma_y$  the static yield stress

$\sigma_{yd}$  the dynamic yield stress

$\dot{\epsilon}$  the strain rate

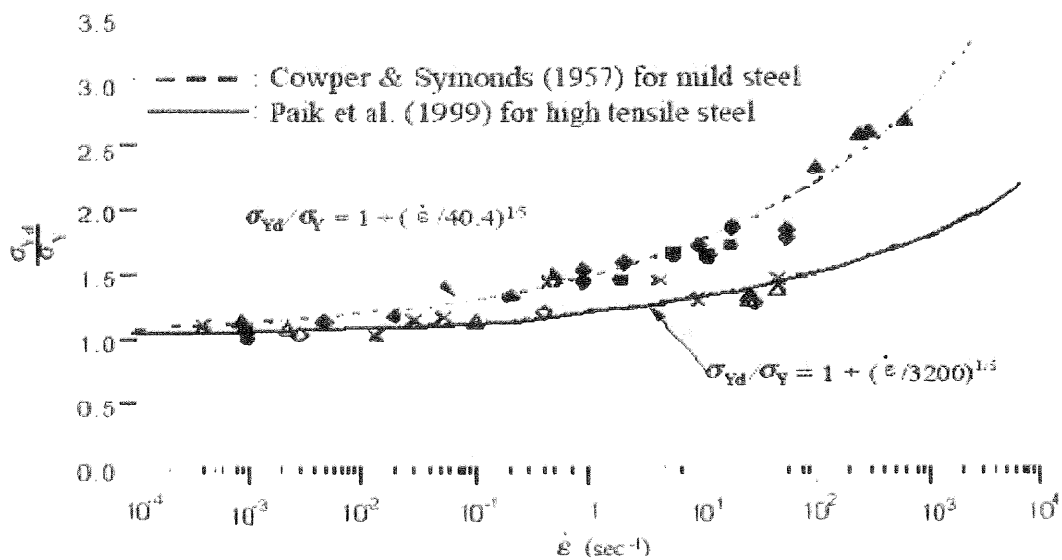


Fig. A. 1 Yield strength versus strain rate for mild and high tensile steels (ISSC 2003)

Figure A.1 plots the Cowper-Symonds formula together with the relevant coefficients for mild or high tensile steels. It is seen from Fig. A.1 that as the strain rate increases the dynamic yield stress increases. It is also found that for higher tensile steel the percentage increase of the ratio of the dynamic yield stress to the static yield stress is smaller than that for mild steel. It is noted that the Cowper-Symonds formula relates to yield stress; the strain rate dependence is smaller for the flow stress at large strains.

The following approximate formula, which is the inverse of the Cowper-Symonds formula for the dynamic yield stress, is then useful for estimating the dynamic fracture strain as a function of the strain rate (Jones 1989), namely

$$\frac{\epsilon_{Fd}}{\epsilon_F} = \left[ 1.0 + \left( \frac{\dot{\epsilon}}{C} \right)^{1/q} \right]^{-1} \quad (\text{A-20})$$

where

$\varepsilon_{Fd}$  the static fracture strain

$\varepsilon_F$  the dynamic fracture strains

#### A4. Shell Element Formulation

Shell elements are the main element type in crashworthiness simulation. The advantage of using 3-d elastic-plastic shell elements in finite element simulation is that the solution includes membrane bending effects, history dependence and also handles contact-friction problem quit easily. Thus the shell element is possible to simulate large complex structures with high accuracy.

The shell elements may differ in shape and the integration rule. Fully integrated four-node quadrilaterals use four integration points in-plane. Under-integrated elements only use one integration point in the plane. In under-integrated shell elements, the strain is evaluated at the element centre and results in the element to exhibit zero energy modes (hourglass modes) where the elements can deform without dissipating energy. Figure A.2 shows the hourglass modes of solid element. Fully integrated elements do not have these hourglass problems, but they may lock in shear which will cause the element to exhibit excessive stiffness.

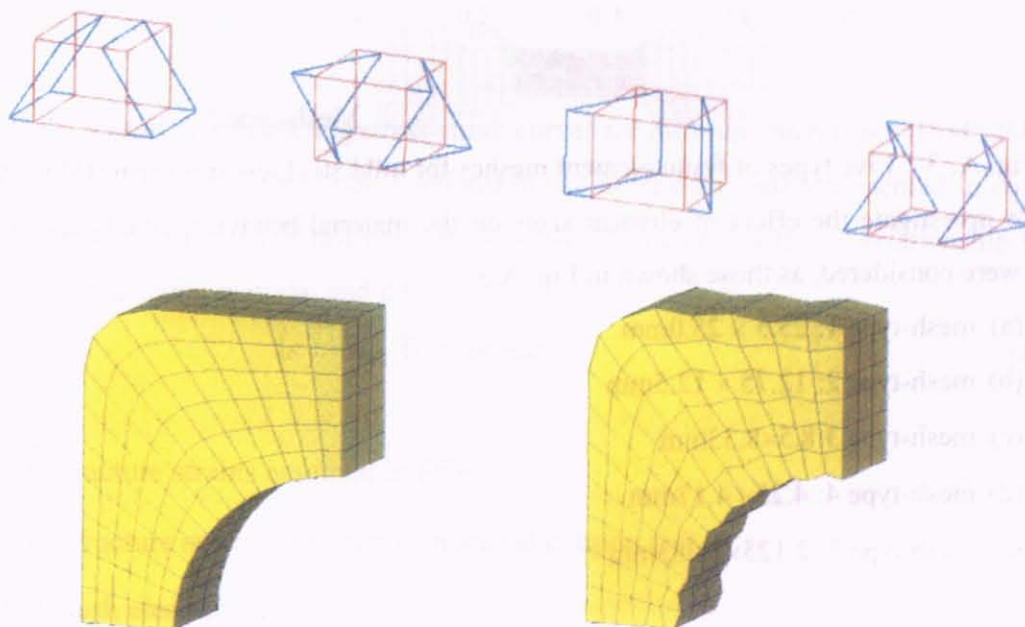


Fig. A. 2 Hourglass modes of solid elements

## A5. Relationship Between the Mesh Size and Rupture Strain

Traditionally, rupture is assumed to occur when the equivalent plastic strain in an analyzed structure reaches a critical value. This critical value, sometime referred to as rupture strain, is related to the strain-stress curves obtained from mechanical tests of uni-axially stretched metal coupons. There has been an interest in defining rupture strain for FEM analyses. The dependency of fracture strain on element size has been studied by Yu (1996), Lehmann and Yu (1998b) and Simonsen and Lauridsen (2000). ISSC (2003) reported a benchmark study on a tensile coupon test of a flat mild steel specimen to investigate the dependency of the input fracture strain on element size.

The stress-strain curve for a tensile coupon test on flat mild steel specimen (thickness of 2mm) in a quasi-static condition was computed using the nonlinear finite element program LS-DYNA. The Belytschko-Tsay type plate-shell elements with a piecewise-linear plasticity material model were employed.

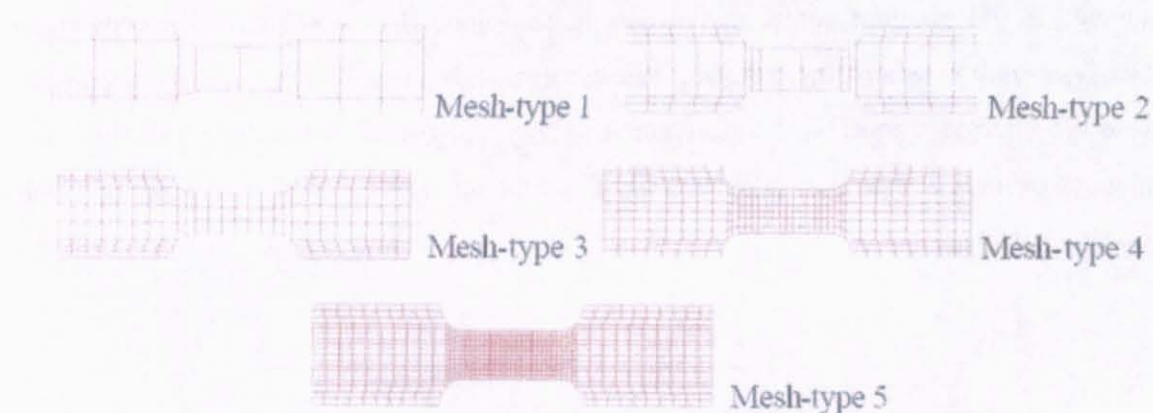


Fig. A. 3 Five types of finite element meshes for mild steel test specimen (ISSC 2003)

To investigate the effect of element sizes on the material behavior, five types of element sizes were considered, as those shown in Fig. A.3.

- (a) mesh-type 1:  $25.5 \times 25.0\text{mm}$
- (b) mesh-type 2:  $12.75 \times 12.5\text{mm}$
- (c) mesh-type 3:  $8.5 \times 8.33\text{mm}$
- (d) mesh-type 4:  $4.25 \times 4.17\text{mm}$
- (e) mesh type 5:  $2.125 \times 2.083\text{mm}$

Figure A.4 compares the finite element solutions with the test in terms of the engineering

stress-strain relationship. It is seen from Fig. A.4 that prediction results with different mesh size overestimates the strain hardening effect. Figure A.5 shows the input fracture strains which were obtained by a trial and error technique so that the “critical” fracture strains by FEA corresponded to those by testing. It is interesting to note that this correspondence was achieved by assuming different level of the fracture strain in the FEA with different meshes.

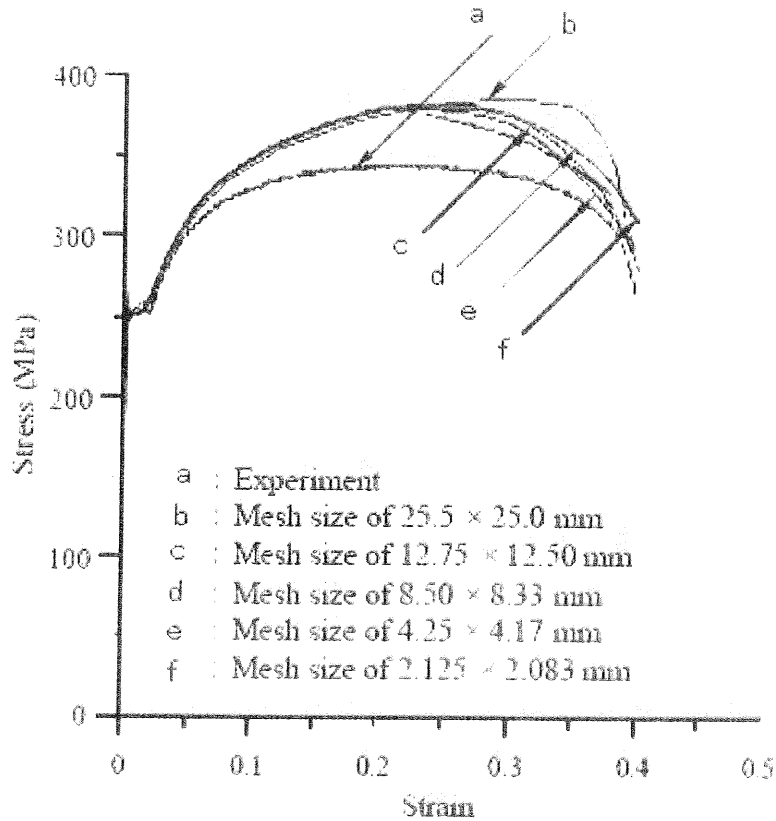


Fig. A. 4 Result of engineering stress-strain curves for different mesh types (ISSC 2003)

The fracture strain – element size relationship can be approximately presented in equation (A-21) to match the findings of this benchmark study. However, a different relationship would be expected for different material and thickness

$$(\varepsilon_f - \varepsilon_{f0})S = \text{constant} \quad (\text{A-21})$$

where

$\varepsilon_f$  the fracture strains assumed in FEA

$\varepsilon_{f0}$  the fracture strains obtained by material coupon test

$S$  the mesh element size

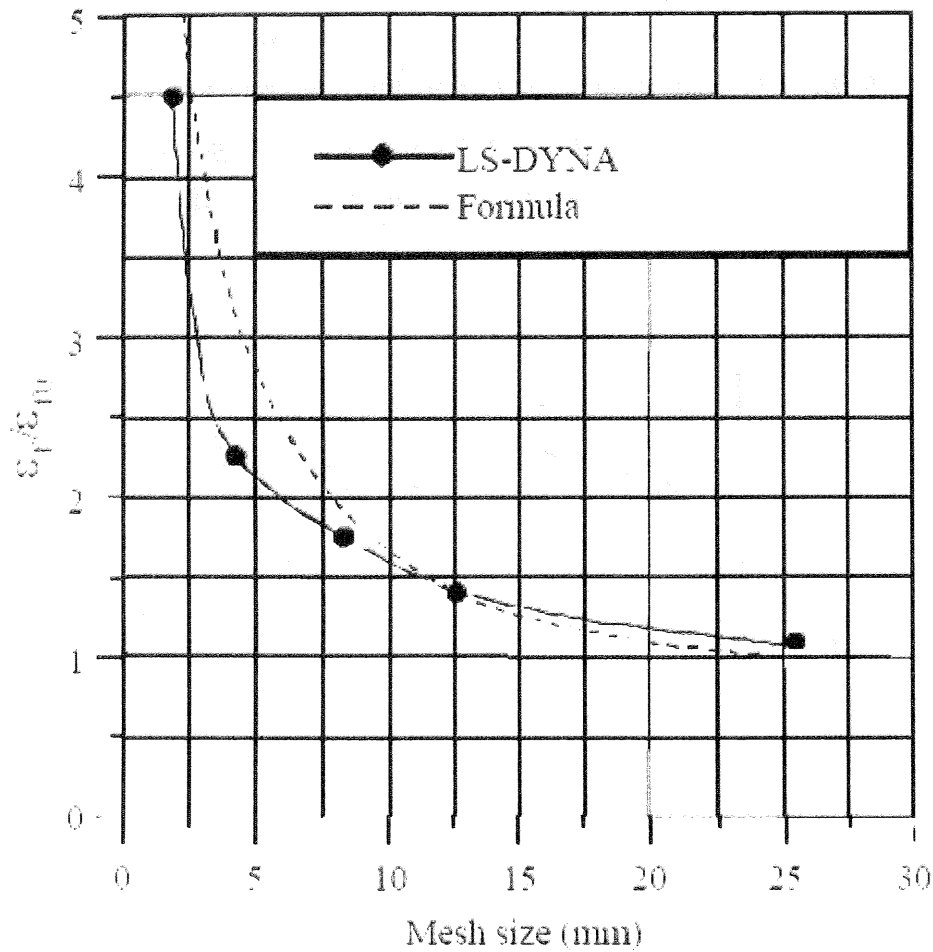


Fig. A. 5 Normalized 'critical' fracture strain VS the finite element size (ISSC 2003)

By comparing tensile tests with numerical simulations they concluded that a fairly small element mesh is required to capture the features of the tensile test.

Ranges of rupture strain were studied by Simonsen and Törnqvist (2004), Okazawa et al., (2004), Yamada et al. (2005), Alsos and Amdahl (2005). Refined simulation of fracture initiation and propagation requires that mesh size are small enough. This in turn makes the analysis of large ship structures very time consuming and computational demanding.

## A6. Arbitrary Lagrangian-Eulerian (ALE) Formulation

In this section, ALE formulation, as implemented in LS-DYNA Keyword Manual 970(2003), is briefly outlined.

In the ALE description, an arbitrary referential coordinate is introduced in addition to the Lagrangian and Eulerian coordinates. The ALE equations are derived by substituting

the relationship between the material time derivative and the reference configuration time derivative,

$$\frac{\partial f(X_i, t)}{\partial t} = \frac{\partial f(x_i, t)}{\partial t} + (v_i - u_i) \frac{\partial f(x_i, t)}{\partial x_i} = \frac{\partial f(x_i, t)}{\partial t} + w_i \frac{\partial f(x_i, t)}{\partial x_i} \quad (\text{A-22})$$

where

$X_i$  the Lagrangian coordinate

$i$  the referential coordinate

$x_i$  the Eulerian coordinate

$v_i$  the material velocity

$u_i$  the mesh velocity

In order to simplify the equations, we introduce the convective velocity  $w_i = v_i - u_i$ .

Thus the governing equations for the ALE formulation are given by:

(i) The conservation of mass equation.

$$\frac{\partial \rho}{\partial t} = -\rho \frac{\partial v}{\partial x_i} - w_i \frac{\partial \rho}{\partial x_i} \quad ((\text{A-23}))$$

(ii) The conservation of momentum equation.

$$\rho \frac{\partial v_i}{\partial t} + \rho w_i \frac{\partial v_i}{\partial x_j} = \sigma_{ij,j} + \rho b_i \quad ((\text{A-24}))$$

(iii) The conservation of total energy equation

$$\rho \frac{\partial E}{\partial t} + \rho w_j \frac{\partial E}{\partial x_j} = \sigma_{ij} v_{i,j} + b_j v_j \quad ((\text{A-25}))$$

where

$\rho$  the density

$b_i$  the body force

$E$  the energy

The overall flow of an ALE time step is as follows

1. perform a Lagrangian time step
2. perform an advection step
  - a) decide which nodes to move
  - b) move the boundary nodes

- c) move the interior nodes
- d) calculate the transport of the element-centred variables
- e) calculate the momentum transport and update the velocity

It should be noted that the cost of the advection step per element is usually much larger than the cost of Lagrangian step. Most of the time in the advection step is spent on calculating the material transported between the adjacent elements, and only a small part of it is spent on calculating how and where the mesh should be adjusted. Second order accurate monotonic advection algorithms are used in LS-DYNA because of their superior coarse mesh accuracy with few elements.

At ALE-Lagrange interface, the Lagrange mesh acts as a moving boundary to the fluid, while fluid pressure is applied to the Lagrange mesh. Using penalty coupling allows us to treat the impact problems in presence of fluid because penalty coupling manages the interactions between a Lagrangian formulation modeling the structure and a ALE formulation modelling the fluid to flow around a structure but not through a structure. Flow through the structure is prevented in an approximate way by applying penalty forces to the fluid and structure. As soon as an ALE node penetrates in a Lagrangian structure, a force of recall is exerted on the contravening node and put its back on the surface of the structure.

More accurately, first, for each Lagrangian node, a search of an ALE element containing this node is done. Then, the penetration depth of each fluid node into the structure surface is evaluated. Finally, penalty forces are determined proportionally to the penetration depth. The coupling method behaves like a spring system.

## A7. Fluid-like Material Model

In commercial FE code LS-DYNA, the Constitutive Model and Equation of State are used to describe fluid-like material model with fluid-like deformation characteristics (air, water, oil etc.).

Constitutive Model (CM) relates  $\sigma'_{ij}$  to  $\dot{\epsilon}'_{ij}$

$$\sigma^v_{ij} = \sigma'_{ij} = \gamma \cdot \dot{\epsilon}'_{ij} \quad (\text{A-26})$$

where

$\dot{\epsilon}'_{ij}$  the deviatoric strain rate (1/s)

$\gamma$  the dynamic viscosity (Pa\*s)

The dynamic viscosity of crude oil is  $1.019 \times 10^{-5} \text{ N/m}^2 \cdot \text{s}$

Equation of State (EOS) relates the pressure ( $P$ ) to the specific rate of change of volume ( $\Delta v/v$ ) of a material at a physical state. For fluid (water, crude oil etc), the EOS is given by:

$$P = K\mu \quad (\text{A-27})$$

where

$$\mu = -\frac{\Delta v}{v} = \frac{\rho}{\rho_0} - 1 \quad (\text{A-28})$$

$\frac{\rho}{\rho_0}$  the ratio of current density to initial density

$K$  the bulk modulus.

The bulk modulus of crude oil is  $1.455 \times 10^9 \text{ N/m}^2$

The total stress in the fluid-like material is

$$\sigma_{ij} = \sigma'_{ij} + \frac{1}{3}\sigma_{kk}\delta_{ij} = \gamma \cdot \dot{\epsilon}'_{ij} - P\delta_{ij} \quad (\text{A-29})$$

## A8. Calculation of Added Mass Coefficients

The hydrodynamic force related to the surge motion of ship is very small compared to sway force. The hydrodynamic forces related to the surge motion cannot be found by strip method. The sectional surge added mass  $A_{1,1}$  related to the forward motion is small compared with the mass of the ship. Professor Motora (1969) found it to be  $(0.02 - 0.07)M$ . A reasonable assumption may be

$$A_{1,1} = 0.05M \quad (\text{A-30})$$

Many researchers have proposed different methods to evaluate the sectional sway added mass coefficients for ship forms; the pioneer works being that by Lewis F. M. (1929), Tasai F. (1961). The sway hydrodynamic added mass coefficient at infinite frequency using Lewis Two Parameter Conformal Mapping Method to be given by

$$A_{2,2}/M = \frac{2\rho}{\pi} \left\{ D^2 + \frac{16}{3} a_3^3 \right\} \quad (\text{A-31})$$

where

$$a_3 = \frac{-c_1 + 3 + (9 - 2c_1)^{1/2}}{c_1} \quad (\text{A-32})$$

$$c_1 = \left[3 + \frac{4\sigma_s}{\pi}\right] + \left[1 - \frac{4\sigma_s}{\pi}\right] [(H_o - 1)/(H_o + 1)]^2 \quad (\text{A-33})$$

$$H_o = \frac{B}{2D} \quad \sigma_s = \frac{A_s}{DB} \quad (\text{A-34})$$

$B$  and  $D$  is the breadth and draught of cross section, respectively

$A_s$  is wetted area of cross section

The added mass coefficient for the pitch/yaw motion of ship,  $A_{5,5} / A_{6,6}$ , is (Pedersen et al., 1993)

$$A_{5,5} = 0.21 \cdot I_{yy} \quad (\text{A-35})$$

$$A_{6,6} = 0.21 \cdot I_{zz} \quad (\text{A-36})$$

## A9. Calculation of Mean Oil Outflow for Side Damage

As the method in BLG 8/18/Add.1 given, the mean oil outflow for side damage shall be calculated as follows:

$$O_{MS} = C_3 \sum_{i=1}^n P_s(i) O_s(i) \quad (m^3) \quad (\text{A-37})$$

where

$O_{MS}$  mean outflow for side damage, in  $m^3$

$i$  represents each cargo tank under consideration

$n$  total number of cargo tanks

$P_s(i)$  the probability of penetrating cargo tank  $i$  from side damage

$O_s(i)$  the outflow from side damage to cargo tank  $i$ , which is assumed equal to the total volume in cargo tank  $i$ , in  $m^3$ ,

$C_3$  0.77 for special VLCC, or 1.0 for all other ships

The probability PS of breaching a compartment from side damage shall be calculated as follows:

$$P_S = P_{SL} \cdot P_{SV} \cdot P_{ST} \quad (\text{A-38})$$

where

$P_{SL} = 1 - P_{Sf} - P_{Sa}$  probability the damage will extend into the longitudinal zone bounded by  $X_a$  and  $X_f$

$P_{SV} = 1 - P_{Su} - P_{Sl}$  probability the damage will extend into the vertical zone bounded by  $Z_l$  and  $Z_u$

$P_{ST} = 1 - P_{Sy}$  probability the damage will extend transversely beyond the boundary defined by  $y$

$P_{Sa}$ ,  $P_{Sf}$ ,  $P_{Sl}$ ,  $P_{Su}$  and  $P_{Sy}$  shall be determined by linear interpolation from the table of probabilities for side damage provided in this regulation.

where

$P_{Sa}$  the probability the damage will lie entirely aft of location  $X_a / L$

$P_{Sf}$  the probability the damage will lie entirely forward of location  $X_f / L$

$P_{Sl}$  the probability the damage will lie entirely below the tank

$P_{Su}$  the probability the damage will lie entirely above the tank

$P_{Sy}$  the probability the damage will lie entirely outboard of the tank

Compartment boundaries  $X_a$ ,  $X_f$ ,  $Z_l$ ,  $Z_u$  and  $y$  shall be developed as follows:

$X_a$  the longitudinal distance from the aft terminal of  $L$  to the aftmost point on the compartment being considered, in metres

$X_f$  the longitudinal distance from the aft terminal of  $L$  to the foremost point on the compartment being considered, in metres

$Z_l$  the vertical distance from the moulded baseline to the lowest point on the compartment being considered, in metres

$Z_u$  the vertical distance from the moulded baseline to the highest point on the compartment being considered, in metres

$y$  the minimum horizontal distance measured at right angles to the

centreline between the compartment under consideration and the side shell in metres

$P_{Sy}$  shall be calculated as follows:

$$P_{Sy} = \begin{cases} \left(24.96 - 199.6 \frac{y}{B}\right) \cdot \frac{y}{B} & \text{for } \frac{y}{B} \leq 0.05 \\ 0.749 + \left[5 - 44.4 \left(\frac{y}{B} - 0.05\right)\right] \left(\frac{y}{B} - 0.05\right) & \text{for } 0.05 < \frac{y}{B} < 0.1 \\ 0.888 + 0.56 \left(\frac{y}{B} - 0.1\right) & \text{for } \frac{y}{B} \geq 0.1 \end{cases} \quad (\text{A-39})$$

$P_{Sy}$  shall not be taken greater than 1.

Table A-1 Probabilities for Side Damage

$X_a / L$	$P_{Sa}$	$X_f / L$	$P_{Sf}$	$Z_l / D$	$P_{Sl}$	$Z_u / D$	$P_{Su}$
0.00	0.000	0.00	0.967	0.00	0.000	0.00	0.968
0.05	0.023	0.05	0.917	0.05	0.000	0.05	0.952
0.10	0.068	0.10	0.867	0.10	0.001	0.10	0.931
0.15	0.117	0.15	0.817	0.15	0.003	0.15	0.905
0.20	0.167	0.20	0.767	0.20	0.007	0.20	0.873
0.25	0.217	0.25	0.717	0.25	0.013	0.25	0.836
0.30	0.267	0.30	0.667	0.30	0.021	0.30	0.789
0.35	0.317	0.35	0.617	0.35	0.034	0.35	0.733
0.40	0.367	0.40	0.567	0.40	0.055	0.40	0.670
0.45	0.417	0.45	0.517	0.45	0.085	0.45	0.599
0.50	0.467	0.50	0.467	0.50	0.123	0.50	0.525
0.55	0.517	0.55	0.417	0.55	0.172	0.55	0.452
0.60	0.567	0.60	0.367	0.60	0.226	0.60	0.383
0.65	0.617	0.65	0.317	0.65	0.285	0.65	0.317
0.70	0.667	0.70	0.267	0.70	0.347	0.70	0.255
0.75	0.717	0.75	0.217	0.75	0.413	0.75	0.197
0.80	0.767	0.80	0.167	0.80	0.482	0.80	0.143
0.85	0.817	0.85	0.117	0.85	0.553	0.85	0.092
0.90	0.867	0.90	0.068	0.90	0.626	0.90	0.046
0.95	0.917	0.95	0.023	0.95	0.700	0.95	0.013
1.00	0.967	1.00	0.000	1.00	0.775	1.00	0.000

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